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Course Description

June 12, 2017 – Fundamentals of Stability for Steel Design: Design of Compression Members

Initially, an overview of flexural, torsional, and flexural-torsional resistance of individual column members will be provided. Emphasis then will be placed on defining and assessing the AISC LRFD and ASD strengths of various structural shapes, including wide flange, round and square HSS, cruciform, equal and unequal single and double leg angles, WT, channel, and built-up shapes.





Learning Objectives

- Explain the design limit states of flexural, torsional, and flexural-torsional buckling resistance of compression members
- Identify the steps to determine the design strengths for column design using the *AISC Specification*.
- Identify the applicable buckling limit states for various structural shapes subjected to compression.
- Understand stability design through the use of design examples.



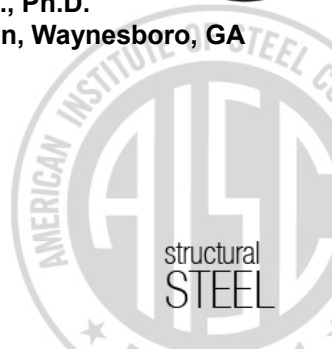
Fundamentals of Stability for Steel Design Session 2: Design of Compression Members

June 12, 2017

There's always a solution in steel.



Presented by
Perry S. Green, P.E., Ph.D.
Bechtel Corporation, Waynesboro, GA

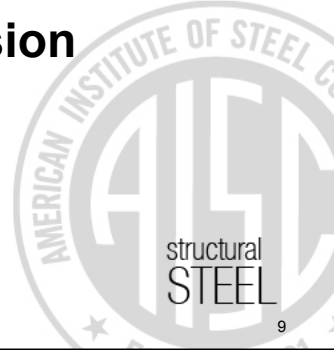


There's always a solution in steel.

Fundamentals of Stability for Steel Design

Session 2 Design of Compression Members

Perry S. Green, P.E., Ph.D.

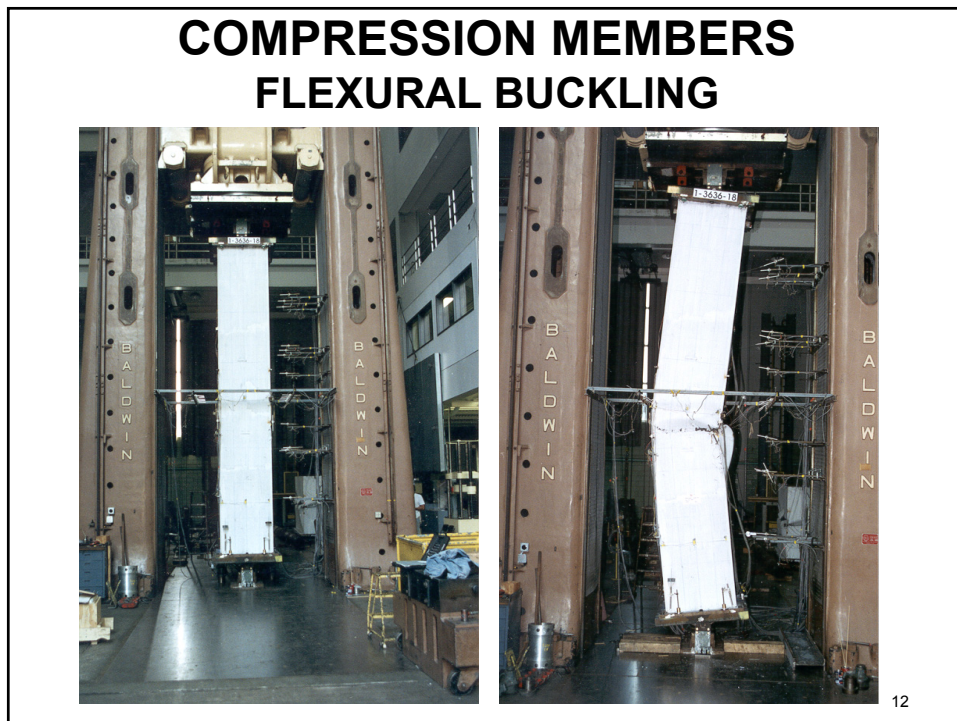


SESSION OUTLINE

- **Introduction**
- **Flexural Buckling**
- **Torsional Buckling**
- **Flexural-Torsional Buckling**
- **Built-Up Members**
- **Design Examples:**
 - Flexural Buckling**
 - Torsional Buckling**
 - Flexural-Torsional Buckling**

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2010 AISC SPECIFICATION CHAPTER E - COMPRESSION MEMBERS

- All compression member strengths are controlled by the limit state of *buckling*

Bad News

There are four different types of buckling

FB: Flexural Buckling

TB: Torsional Buckling

FTB: Flexural-Torsional Buckling

LB: Local Plate Buckling

Good News

Not all types are applicable to every member

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ANSI/AISC 360-10 *Specification for the Design of Steel Buildings, 2010*

CHAPTER E

DESIGN OF MEMBERS FOR COMPRESSION

- E1. GENERAL PROVISIONS
- E2. EFFECTIVE LENGTH
- E3. FLEXURAL BUCKLING OF MEMBERS WITHOUT SLENDER ELEMENTS**
- E4. TORSIONAL AND FLEXURAL-TORSIONAL BUCKLING OF MEMBERS WITHOUT SLENDER ELEMENTS**
- E5. SINGLE ANGLE COMPRESSION MEMBERS**
- E6. BUILT-UP MEMBERS**
- E7. MEMBERS WITH SLENDER ELEMENTS

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2010 AISC SPECIFICATION CHAPTER E - COMPRESSION MEMBERS

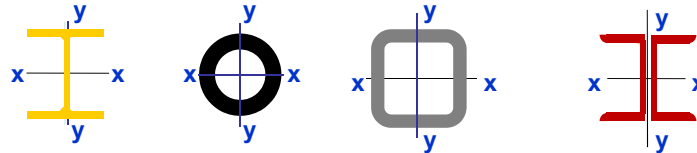
FB: Flexural Buckling
TB: Torsional Buckling
FTB: Flexural-Torsional Buckling
~~**LB: Local Plate Buckling**~~

only for members with
 slender plate elements

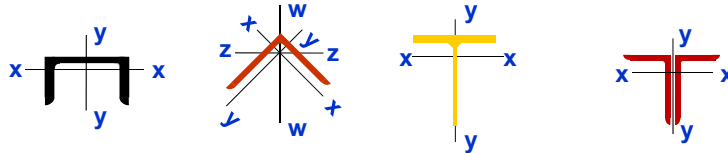
Cross Section	Without Slender Elements	
	Sections in Chapter E	Limit States
	E3 E4	FB TB
	E3 E4	FB FTB
	E3	FB
	E3	FB
	E3 E4	FB FTB
	E3 E4 E6	FB FTB
	E5	FB TB
	E3	FB
Unsymmetrical shapes other than single angles	E4	FTB

TYPES OF COMPRESSION MEMBERS

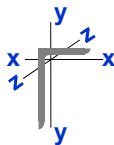
- Doubly-symmetric Members**



- Singly-symmetric Members**



- Unsymmetric Members**



FLEXURAL BUCKLING DOUBLY-SYMMETRIC MEMBERS

Most cases in design of W Shapes involve flexural buckling

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2010 AISC SPECIFICATION CHAPTER E - COMPRESSION MEMBERS

CROSS SECTION	WITHOUT SLENDER ELEMENTS	
	SECTIONS IN CHAPTER E	LIMIT STATES
	E3 E4	FB TB

FB: Flexural Buckling
 TB: Torsional Buckling
 FTB: Flexural-Torsional Buckling

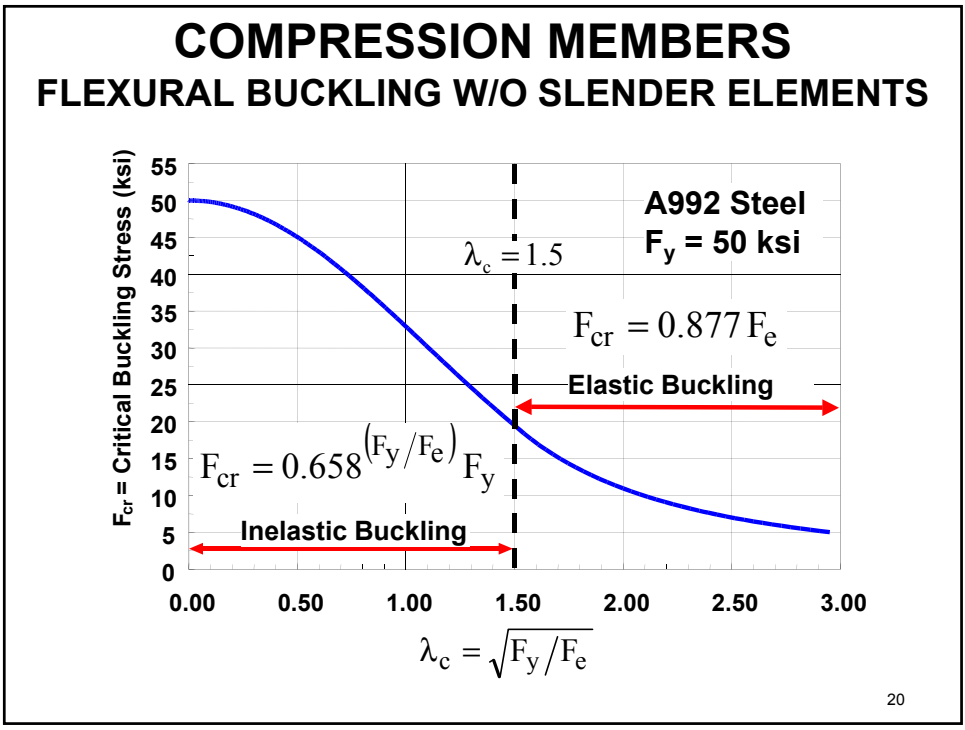
For doubly-symmetric I shapes *flexural buckling* will always control if twist of the member is prevented at all supports and brace points, *i.e.*, the lateral unbraced length \geq torsional unbraced length

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INEFFECTIVE TORSIONAL SUPPORT

Torsional Buckling will be discussed later

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COMPRESSION MEMBERS
FLEXURAL BUCKLING W/O SLENDER ELEMENTS
ANSI/AISC 360-10 Specification for the Design of Steel Buildings, 2010

$\frac{KL}{r} \leq 4.71 \sqrt{\frac{E}{F_y}}$
 or
 $\frac{F_y}{F_e} \leq 2.25$

NO

YES

$F_{cr} = \left[0.658 \frac{F_y}{F_e} \right] F_y$



$P_n = F_{cr} A_g$ (E3-1)

$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2}$ (E3-4)

$F_{cr} = 0.877 F_e$ (E3-3)

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COMPRESSION MEMBER
DESIGN EXAMPLE 1
ANSI/AISC 360-10 Specification for the Design of Steel Buildings, 2010

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WF MEMBER DESIGN STRENGTH

$P_D = 125$ kips
 $P_L = 400$ kips
 $P_a = 525$ kips
 $P_u = 790$ kips

20'

NOTE: From the Column Strength Tables $\phi_c P_n = 816$ kips

Check Required Strength W12 x 96

$A_g = 28.2$ in.² $I_y = 270$ in.⁴ $r_y = 3.09$ in.
 $F_y = 50$ ksi $KL_y = 240$ in.

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = \frac{\pi^2 (29000)}{\left(\frac{240}{3.09}\right)^2} = 47.44 \text{ ksi} \quad (E3-4)$$

$$\frac{F_y}{F_e} = \frac{50}{47.44} = 1.05 \leq 2.25 \quad \therefore \text{Inelastic}$$

$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}}\right] F_y = \left[0.658^{1.05}\right] (50) = 32.17 \text{ ksi} \quad (E3-2)$$

$$P_n = F_{cr} A_g = (32.17)(28.2) = 907.0 \text{ kips} \quad (E3-1)$$

$$\frac{P_n}{\Omega_c} = \frac{907}{1.67} = 543 \text{ kips} > 525 \text{ kips} \quad \therefore \text{OK}$$

$$\phi_c P_n = 0.9(907) = 816 \text{ kips} > 790 \text{ kips} \quad \therefore \text{OK}$$

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POTENTIAL BUCKLING MODES

Flexural Buckling


Section A-A

Torsional Buckling

Section B-B

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2010 AISC SPECIFICATION CHAPTER E - COMPRESSION MEMBERS

CROSS SECTION	WITHOUT SLENDER ELEMENTS	
	SECTIONS IN CHAPTER E	LIMIT STATES
	E3 E4	FB TB

FB: Flexural Buckling

TB: Torsional Buckling

FTB: Flexural-Torsional Buckling

Torsional buckling occurs over short lengths

- Usually KL/r will be less than approximately 50
- Doubly-symmetric section
- Wide flange, cruciform, and point symmetric sections

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TORSIONAL BUCKLING OPEN CROSS SECTION MEMBERS

- **Buckling in a torsional mode (torsion of non-circular sections involves torsional shear and warping:**
 - K_z is normally taken as 1.0
 - Effective length in torsion is $K_z L_z = L_z$
 - Function of the torsional constant $J \approx \Sigma(bt^3/3)$
 - Capacity of section with warping stiffness is a function of warping constant C_w
 - C_w, J, r_x, r_y are given in the Properties Tables where x and y are the axes of symmetry of the section
 - $E = 29,000$ ksi, $G = 0.385E$ (11,200 ksi)

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COMPRESSION MEMBERS
TORSIONAL BUCKLING W/O SLENDER ELEMENTS
ANSI/AISC 360-10 Specification for the Design of Steel Buildings, 2010

$$\frac{F_y}{F_e} \leq 2.25$$

YES

$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}} \right] F_y \quad (E3-2)$$

NO



$$F_{cr} = 0.877 F_e \quad (E3-3)$$

$$P_n = F_{cr} A_g \quad (E4-1)$$

$$F_e = \left[\frac{\pi^2 E C_w}{(K_z L)^2} + GJ \right] \frac{1}{I_x + I_y} \quad (E4-4)$$

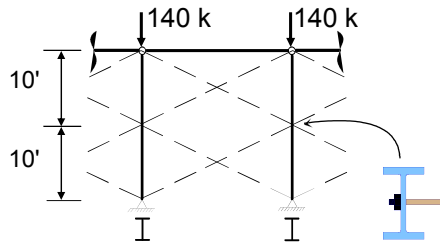
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COMPRESSION MEMBER
DESIGN EXAMPLE 2
ANSI/AISC 360-10 Specification for the Design of Steel Buildings, 2010

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TORSIONAL BUCKLING EXAMPLE



Are the columns stable with the 140 k factored load if bracing at midheight is supplied by tie rods?

Columns: W16x26; $F_y = 50$ ksi; $P_y = 384$ kips

$$I_y = 9.59 \text{ in}^4; J = 0.26 \text{ in}^4; A_g = 7.68 \text{ in}^2$$

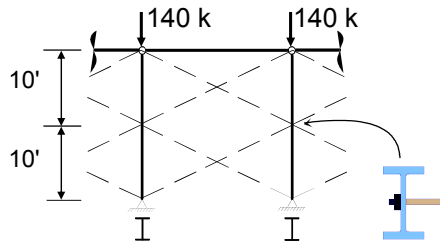
$$I_x = 301 \text{ in}^4; C_w = 565 \text{ in}^6$$

Flexural Buckling Effective Length: $L_y = 10$ ft

$$\phi P_n = 150 \text{ kips} > 140 \text{ kips} \quad \therefore \text{OK}$$

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TORSIONAL BUCKLING EXAMPLE



Torsional Buckling
Effective Length: $L_T = 20$ ft.

$$F_e = \left[\frac{\pi^2 (29000)(565)}{(20 \times 12)^2} + 11200(0.262) \right] \frac{1}{(301 + 9.59)} = 18.5 \text{ ksi} \quad (\text{E4-4})$$

$$\frac{F_y}{F_e} = \frac{50}{18.5} = 2.70 > 2.25 \quad \therefore \text{Elastic} \quad F_{cr} = 0.877(18.5) = 16.2 \text{ ksi} \quad (\text{E3-3})$$

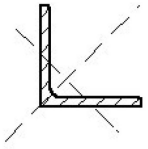
$$P_{Tn} = F_{cr} A_g = (16.2)(7.68) = 124.4 \text{ kips} \quad (\text{E4-1})$$

$$\phi_c P_{Tn} = 0.9(124.4) = 112 \text{ kips} < 140 \text{ kips} \quad \therefore \text{NG}$$

Note: If $L_T = 10$ ft., $P_T = 283$ k

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2010 AISC SPECIFICATION CHAPTER E - COMPRESSION MEMBERS

CROSS SECTION	WITHOUT SLENDER ELEMENTS	
	SECTIONS IN CHAPTER E	LIMIT STATES
	E5	<div style="border: 1px solid red; border-radius: 50%; padding: 5px; display: inline-block;"> FB TB </div>

FB: Flexural Buckling
TB: Torsional Buckling
FTB: Flexural-Torsional Buckling

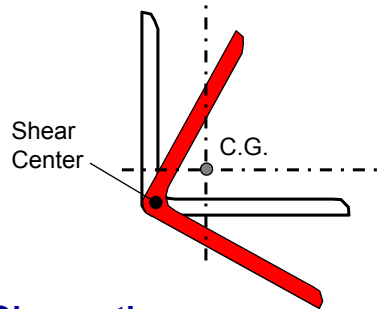
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TORSIONAL BUCKLING EQUAL LEG ANGLE MEMBERS

$$P_{crz} = \frac{GJA}{I_p}$$

$$A = 2bt, \quad J = 2 \frac{bt^3}{3}$$

$$I_p = 2 \frac{b^3t}{3}, \quad C_w \approx 0$$



Observations:

Torsional buckling is not a function of the angle length

The buckling capacity decreases for large leg sizes

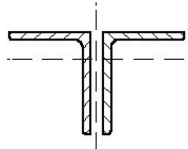
32

TORSIONAL BUCKLING EQUAL LEG ANGLE MEMBERS



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2010 AISC SPECIFICATION CHAPTER E - COMPRESSION MEMBERS

CROSS SECTION	WITHOUT SLENDER ELEMENTS	
	SECTIONS IN CHAPTER E	LIMIT STATES
	E3 E4 E6	FB FTB

FB: Flexural Buckling
TB: Torsional Buckling
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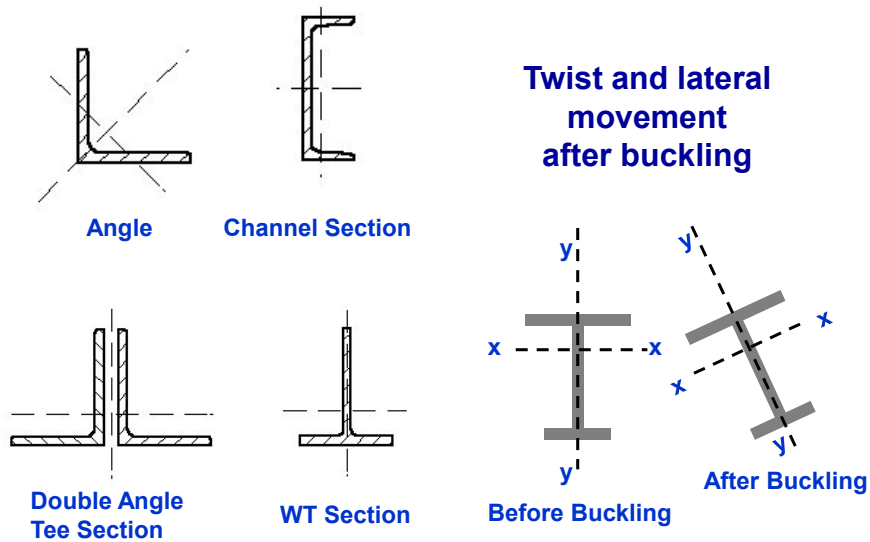
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FLEXURAL-TORSIONAL BUCKLING SINGLY AND UNSYMMETRIC MEMBERS

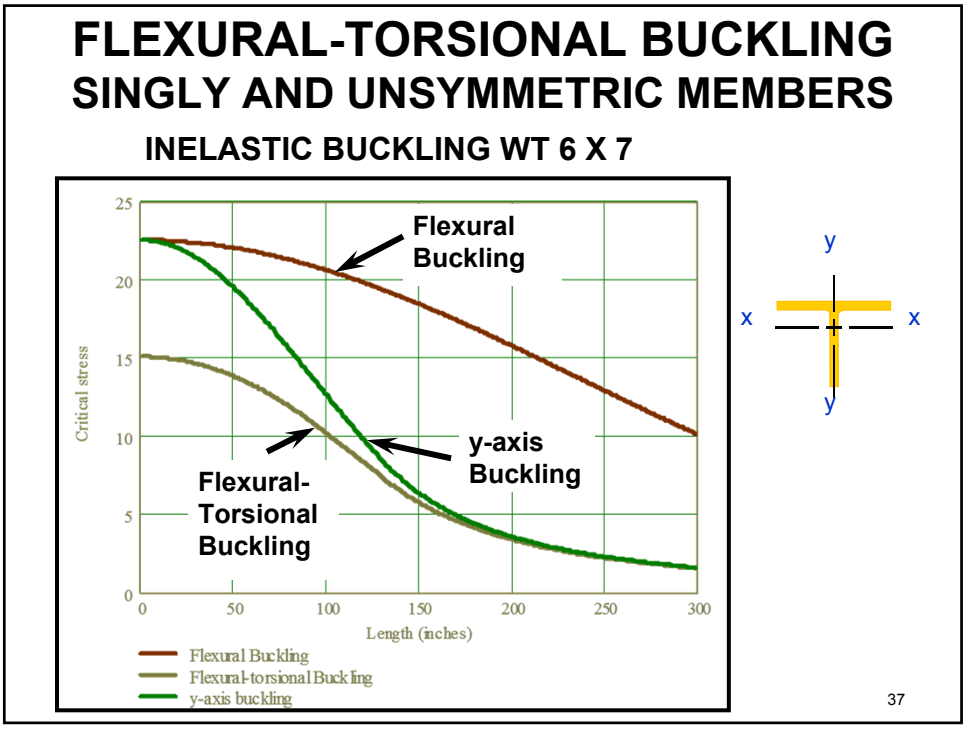
- Flexural buckling will occur in the plane of symmetry
- Sections always rotate about the shear center; the shear center lies on the axis of symmetry
- Singly symmetric sections can buckle either in a Flexural or Flexural-Torsional mode
- Flexural-Torsional buckling will occur by twisting about a point in the plane of symmetry
- Unsymmetric sections will always buckle in a Flexural-Torsional mode

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FLEXURAL-TORSIONAL BUCKLING SINGLY AND UNSYMMETRIC MEMBERS



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POLLING QUESTION 1

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COMPRESSION MEMBERS BUCKLING OF SINGLE ANGLES WITH OR W/O SLENDER ELEMENTS

AISC/LRFD Specification for Single Angle Members (Nov. 10, 2000) permitted design for L/r_z only. OK for usual dimensions, but can be unconservative for short single angles

The 2005 AISC Specification recommended using effective length formulas from ASCE Transmission Line Standard for single angle columns attached to a gusset plate to one leg

See the SSRC Guide to Stability Design Criteria for Metal Structures Chapter 11 for more info

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COMPRESSION MEMBERS BUCKLING OF SINGLE ANGLES WITH OR W/O SLENDER ELEMENTS

- The effects of eccentricity on single angle members are permitted to be neglected when the members are evaluated as axially loaded compression members using one of the effective slenderness ratios specified below, provided that:
 - Members are loaded at the ends in compression through the same one leg
 - Members are attached by welding or by minimum two-bolt connections; and
 - No intermediate transverse loads are present

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COMPRESSION MEMBERS EQUIVALENT KL/r FOR SINGLE ANGLES

Case 1 - E5(a): Planar truss, attachment to chord or gusset plate

$\frac{L}{r_x} \leq 80$

YES

$\frac{KL}{r} = 72 + 0.75 \frac{L}{r_x}$ (E5-1)

NO

$\frac{KL}{r} = 32 + 1.25 \frac{L}{r_x}$ (E5-2)

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COMPRESSION MEMBERS EQUIVALENT KL/r FOR SINGLE ANGLES

Additional requirements for unequal leg angles:

$$\left(\frac{KL}{r} \right)_{eq} \geq 0.95 \frac{L}{r_z}$$

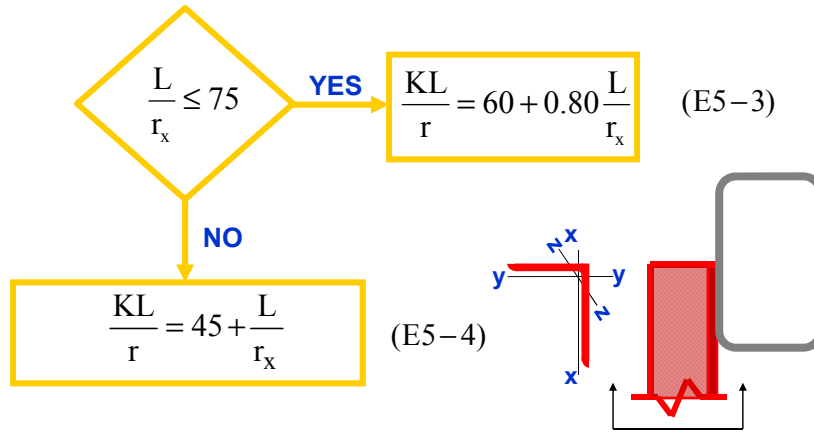
**For unequal leg angles with $b_{long}/b_{short} < 1.7$
 where attachment is to the shorter leg, increase**

$$\left(\frac{KL}{r} \right)_{eq} \text{ by } 4 \left[\left(\frac{b_{long}}{b_{short}} \right)^2 - 1 \right]$$

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COMPRESSION MEMBERS EQUIVALENT KL/r FOR SINGLE ANGLES

Case 2 - E5(b): Box or space truss



COMPRESSION MEMBERS EQUIVALENT KL/r FOR SINGLE ANGLES

Additional requirements for unequal leg angles:

$$\left(\frac{KL}{r}\right)_{eq} \geq 0.82 \frac{L}{r_z}$$

For unequal leg angles with $b_{long}/b_{short} < 1.7$
 where attachment is to the shorter leg, increase

$$\left(\frac{KL}{r}\right)_{eq} \text{ by } 6 \left[\left(\frac{b_{long}}{b_{short}}\right)^2 - 1 \right]$$

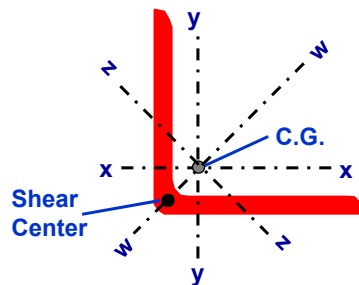
COMPRESSION MEMBER DESIGN EXAMPLE 3

*ANSI/AISC 360-10 Specification
for the Design of Steel Buildings, 2010*



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MEMBER DESIGN STRENGTH SINGLE ANGLE COLUMN $\angle 2 \times 2 \times 3/16$



An equal leg angle is welded to the web of tee chords of a planar truss. Is it more advantageous to use 50 ksi material rather than 36 ksi yield strength material?

Section Properties:

$$A = 0.722 \text{ in.}^2$$

$$r_x = 0.612 \text{ in.}$$

$$r_z = 0.389 \text{ in.}$$

$$Q_s = 1.0$$

Compression Member

$$\text{Length: } L = 44 \text{ in.}$$

Material Properties:

$$F_y = 36 \text{ ksi or } 50 \text{ ksi}$$

$$E = 29,000 \text{ ksi}$$

$$G = 0.385 E$$

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MEMBER DESIGN STRENGTH SINGLE ANGLE COLUMN \perp 2 x 2 x 3/16

Effective Slenderness

$$\frac{b}{t} = \frac{2}{0.1875} = 10.7 < 20$$

$$\frac{L}{r_x} = \frac{44}{0.612} = 72 < 80$$

$$\left(\frac{KL}{r}\right)_x = 72 + 0.75 \frac{L}{r_x} = 126 \leq 4.71 \sqrt{\frac{E}{F_y}} = 133 \quad (E5-1)$$

Details:

Equal leg angle

One leg connected to tee chord

Attached by welding

Width-to-thickness ratio meets
criteria in Section E5

$F_y = 36$ ksi

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MEMBER DESIGN STRENGTH SINGLE ANGLE COLUMN \perp 2 x 2 x 3/16

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = 18.03 \text{ ksi} \quad (E3-4)$$

$$F_{cr} = \left[0.658 \frac{F_y}{F_e} \right] F_y = 15.60 \text{ ksi} \quad (E3-2)$$

$$P_n = F_{cr} A_g = 11.27 \text{ kips} \quad (E3-1)$$


 ASD $\frac{P_n}{\Omega} = 6.75$ kips LRFD $\phi P_n = 10.14$ kips

Single angle will buckle inelastically in a flexural mode

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MEMBER DESIGN STRENGTH SINGLE ANGLE COLUMN \perp 2 x 2 x 3/16

Effective Slenderness

$$\frac{b}{t} = \frac{2}{0.1875} = 10.7 < 20$$

Details:
All same except
 $F_y = 50$ ksi

$$\frac{L}{r_x} = \frac{44}{0.612} = 72 < 80$$

$$\left(\frac{KL}{r}\right)_x = 72 + 0.75 \frac{L}{r_x} = 126 > 4.71 \sqrt{\frac{E}{F_y}} = 113 \quad (E5-1)$$

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MEMBER DESIGN STRENGTH SINGLE ANGLE COLUMN \perp 2 x 2 x 3/16

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = 18.03 \text{ ksi} \quad (E3-4)$$

$$F_{cr} = 0.877 F_e = 15.81 \text{ ksi} \quad (E3-2)$$

$$P_n = F_{cr} A_g = 11.41 \text{ kips} \quad (E3-1)$$

➔
 ASD $\frac{P_n}{\Omega} = 6.83 \text{ kips}$

 LRFD $\phi P_n = 10.27 \text{ kips}$

Single angle will buckle elastically in a flexural mode

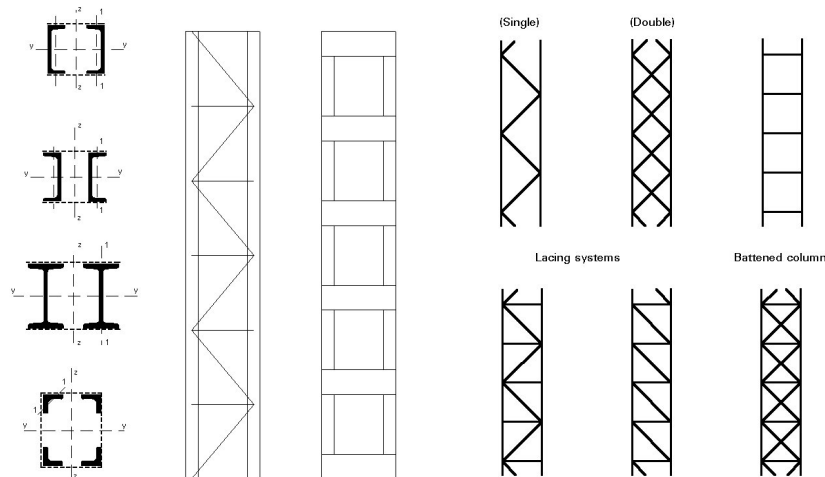
NOTE: Design strength has only increased about 1%

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COMPRESSION MEMBERS BUILT-UP COLUMNS

- **Built-up members are comprised of two or more sections**
 - Stitch bolts
 - Batten plates
 - Lacing
 - Combined batten plates and lacing
 - Perforated cover plates
- **Built-up batten member buckling is somewhat similar to frame buckling**
 - Battens act like beams
 - Battens get shear and moment due to the bending of the frame like built-up member at the time of buckling

COMPRESSION MEMBERS BUILT-UP LACED OR BATTEN COLUMNS



*See the SSRC Guide to Stability Design Criteria
for Metal Structures Chapter 3 for more info*

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COMPRESSION MEMBERS BUCKLING OF BUILT-UP MEMBERS COMPOSED OF TWO SHAPES

(a) For intermediate connectors that are bolted snug-tight

$$\left(\frac{KL}{r}\right)_m = \sqrt{\left(\frac{KL}{r}\right)_o^2 + \left(\frac{a}{r_i}\right)^2} \quad (E6-1)$$

a = distance between connectors (in.)

r_i = minimum radius of gyration of individual component (in.)

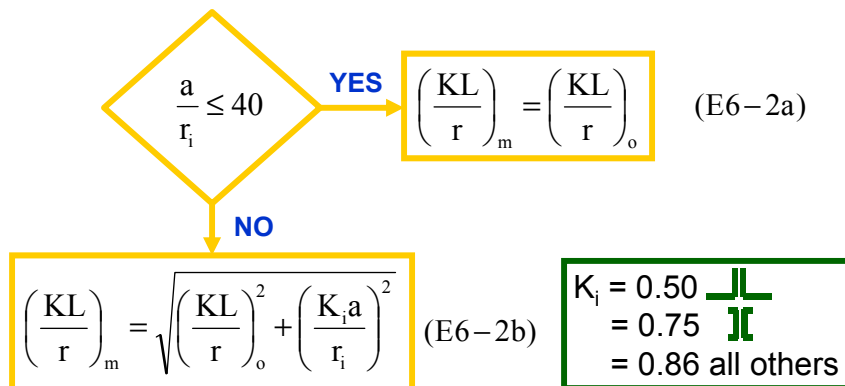
$\left(\frac{KL}{r}\right)_m$ = modified slenderness of built-up member

$\left(\frac{KL}{r}\right)_o$ = slenderness ratio of built-up member acting as a unit in the buckling direction being considered

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COMPRESSION MEMBERS BUCKLING OF BUILT-UP MEMBERS COMPOSED OF TWO SHAPES

(b) For intermediate connectors that are welded or are connected by means of pretensioned bolts



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COMPRESSION MEMBER DESIGN EXAMPLE 4

*ANSI/AISC 360-10 Specification
for the Design of Steel Buildings, 2010*

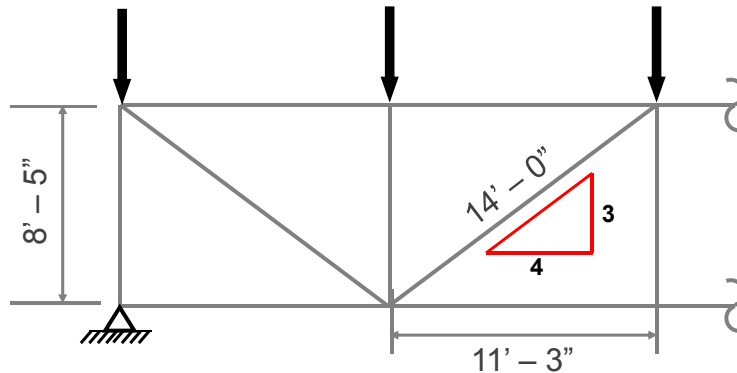


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MEMBER DESIGN STRENGTH DIAGONAL BRACE 2L s 6 x 4 x 5/8 LLBB

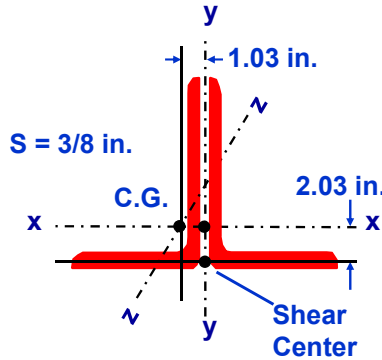
What is the design strength of the 14 ft. long diagonal brace that is comprised of 2L's 6 x 4 x 5/8 LLBB, A36 steel.

There are two intermediate bolted connectors at a spacing of 56 in.
Use $K = 1.0$



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MEMBER DESIGN STRENGTH DIAGONAL BRACE 2L s 6 x 4 x 5/8 LLBB



Material Properties

$F_y = 36$ ksi
 $E = 29,000$ ksi
 $G = 0.385 E$

Single Angle Section Properties

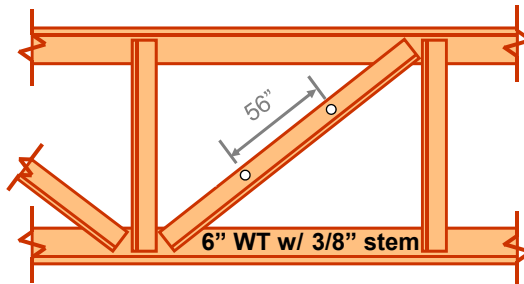
$J = 0.775$ in.⁴ $C_w = 1.59$ in.⁶
 $r_z = 0.859$ in. $r_{yi} = 1.13$ in.

Double Angle Section Properties

$A = 11.7$ in.² $I_x = 42.0$ in.⁴
 $r_x = 1.89$ in. $r_y = 1.66$ in.
 $\bar{r}_0 = 3.05$ in. $J = 1.55$ in.⁴
 $H = 0.684$ in.

57

MEMBER SLENDERNESS DIAGONAL BRACE 2L s 6 x 4 x 5/8 LLBB



x - x axis: $KL/r_x = 88.9$
z - z axis: $a/r_z = 65.2$
y - y axis: $KL/r_y = 101.2$

The intermediate connectors are only snug-tight
Determine the design strength neglecting flexural-torsional buckling;
Y-Y axis is built-up and acts as a single unit, therefore,

$$\left(\frac{KL}{r}\right)_m = \sqrt{\left(\frac{KL}{r}\right)_o^2 + \left(\frac{a}{r_i}\right)^2} \quad (E6-1)$$

58

MEMBER DESIGN STRENGTH DIAGONAL BRACE 2L s 6 x 4 x 5/8 LLBB

$$\left(\frac{KL}{r}\right)_m = \sqrt{\left(\frac{KL}{r}\right)_o^2 + \left(\frac{a}{r_i}\right)^2} = \sqrt{(101.2)^2 + (65.2)^2} = 120.38$$

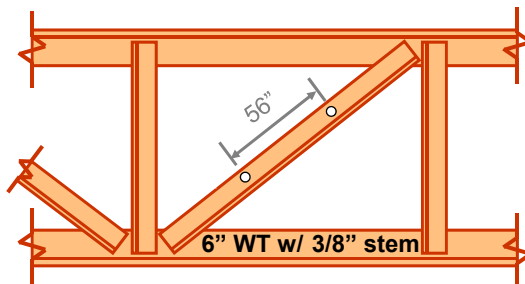
$$F_{ey} = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)_m^2} = 19.75 \text{ ksi} \quad F_y/F_{ey} = 36/19.75 = 1.82 < 2.25$$

$$F_{cry} = 0.658^{F_y/F_{ey}} F_y = 16.79 \text{ ksi}$$

$$P_n = A_g F_{cr} = (11.7)(16.79) = 196.4 \text{ kips}$$

59

MEMBER SLENDERNESS DIAGONAL BRACE 2L s 6 x 4 x 5/8 LLBB



x - x axis: $KL/r_x = 88.9$

z - z axis: $a/r_z = 65.2$

y - y axis: $KL/r_y = 101.2$

$$\frac{a}{r_i} = \frac{56}{0.859} = 65.2 > 40$$

The intermediate connectors are pretensioned
Determine the design strength neglecting flexural-torsional buckling;

Y-Y axis is built-up and acts as a single unit, therefore,

$$\left(\frac{KL}{r}\right)_m = \sqrt{\left(\frac{KL}{r}\right)_o^2 + \left(\frac{K_i a}{r_i}\right)^2} \quad (E6-2b)$$

60

MEMBER DESIGN STRENGTH DIAGONAL BRACE 2L s 6 x 4 x 5/8 LLBB

$$\left(\frac{KL}{r}\right)_m = \sqrt{\left(\frac{KL}{r}\right)_o^2 + \left(\frac{K_i a}{r_i}\right)^2}$$

$$= \sqrt{(101.2)^2 + [(0.5)(65.2)]^2} = 106.3$$

$$F_{ey} = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)_y^2} = 25.32 \text{ ksi} \quad F_y/F_{ey} = 36/25.32 = 1.42 < 2.25$$

$$F_{cry} = 0.658 F_e \quad F_y = 19.85 \text{ ksi}$$

$$P_n = A_g F_{cr} = (11.7)(19.85) = 232.3 \text{ kips} \quad \leftarrow \text{Use}$$

61

DETERMINATION OF CRITICAL STRESS DIAGONAL BRACE 2L s 6 x 4 x 5/8 LLBB

Flexural-Torsional Buckling

$$F_{crz} = \frac{\left(\frac{\pi^2 EC_w}{(K_z L)^2} + GJ\right)}{A_g \bar{r}_o^2} \cong \frac{GJ}{A_g \bar{r}_o^2} \quad (E4-3)$$

Small compared to GJ, therefore this term can be safely ignored for angles, double angles and tees.
In this case, 32 versus 17360

$$F_{crz} = \frac{GJ}{A_g \bar{r}_o^2} = \frac{(11200)(1.55)}{(11.7)(3.05)^2} = 159.5 \text{ ksi}$$

$$F_{cr} = \frac{F_{cry} + F_{crz}}{2H} \left[1 - \sqrt{1 - \frac{4F_{cry}F_{crz}H}{(F_{cry} + F_{crz})^2}} \right] \quad (E4-2)$$

62

MEMBER DESIGN STRENGTH DIAGONAL BRACE 2L s 6 x 4 x 5/8 LLBB

$$F_{cr} = \frac{19.85 + 159.5}{2(0.684)} \left[1 - \sqrt{1 - \frac{4(19.85)(159.5)(0.684)}{(19.85 + 159.5)^2}} \right]$$

$$F_{cr} = 19.0 \text{ ksi}$$

$$P_n = A_g F_{cr} = (11.7)(19.0) = 222.4 \text{ kips}$$

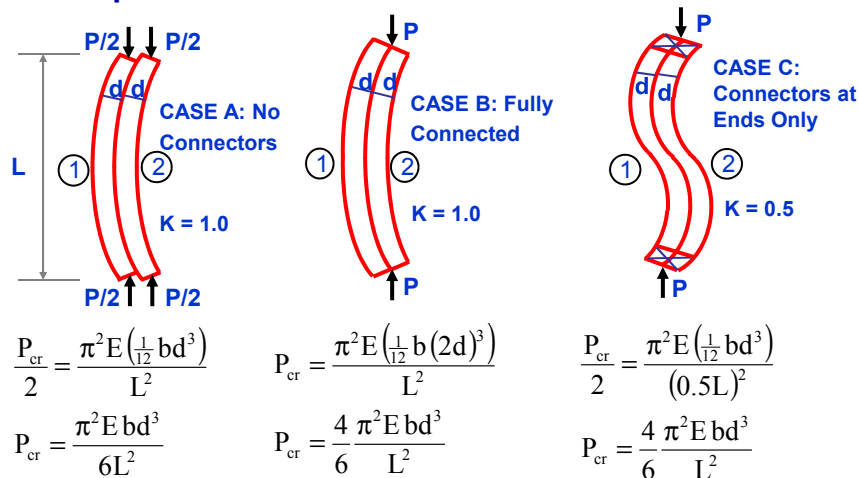
$$\text{ASD } \frac{P_n}{\Omega} = 133.2 \text{ kips} \quad \text{LRFD } \phi P_n = 200.10 \text{ kips}$$

The Flexural-Torsional buckling load is ~5% less than the Euler buckling load in this case. If the legs of the angles or stem of the tees are thin, Flexural-Torsional buckling could be up to 20% less than the Flexural buckling load.

63

COMPRESSION MEMBERS BUILT-UP COLUMNS

Effect of Connector End Restraint on the Behavior of Built-up Columns - Consider Three Cases:



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COMPRESSION MEMBERS BUILT-UP COLUMNS

CASE A: With no connectors, I_{eff} of the cross-section is the sum of the I of the individual components,

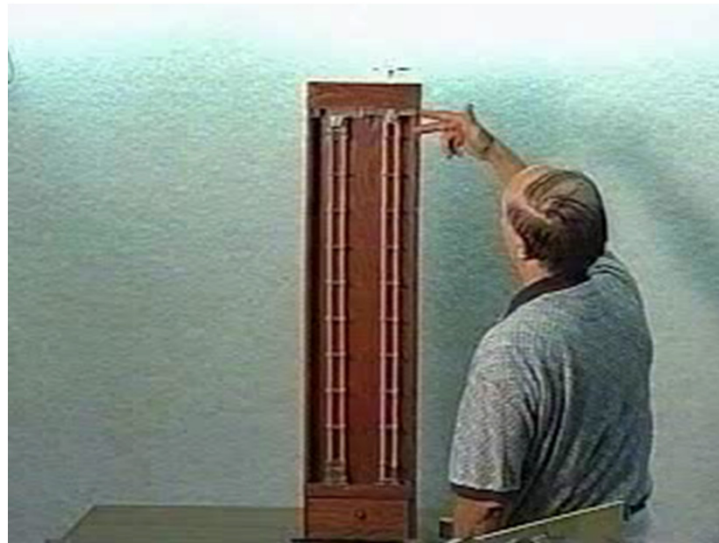
$$I_{eff} = 2 \left[\frac{1}{12} b d^3 \right].$$

CASE B: With the two components fully connected, no slip occurs between members, the cross-section acts as a single unit and $I_{eff} = \frac{1}{12} b(2d)^3$ or four times that with no connectors.

CASE C: Providing just a few connectors will give an effective member stiffness somewhere between these two extremes. The maximum effect will occur when connectors are placed near the ends where the relative displacement between the contact surfaces would be a maximum. Connections near midspan will have no effect.

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COMPRESSION MEMBERS BUILT-UP COLUMNS



66

COMPRESSION MEMBERS BUILT-UP COLUMNS

Shear Connector Design Requirements for Slip

Total Shear Force = $P\Delta \frac{Q}{I_y}$

$\Delta = \frac{\Delta_0}{1 - \frac{P}{P_{cr}}}$ where $\Delta_0 = 0.001L$ and $P/P_{cr} = 0.877$, then

Required Slip Resistance = $0.008P_u L \frac{Q}{I_y}$

67

COMPRESSION MEMBERS BUILT-UP COLUMNS

Shear Force Distribution Along Length of the Column

With 2 intermediate fasteners along the length of the column, connector "a" can only provide 13% of the shear transfer. The end connection must be designed for 87% of the total shear force.

YOU CANNOT DISTRIBUTE THE SHEAR CONNECTORS UNIFORMLY AS IN COMPOSITE BEAMS

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COMPRESSION MEMBER DESIGN EXAMPLE 5

*ANSI/AISC 360-10 Specification for the Design of Steel
Buildings, 2010*



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COMPRESSION MEMBERS BUILT-UP COLUMNS

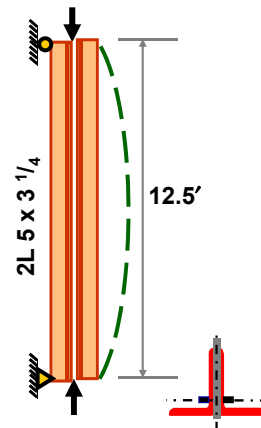
**Design Slip Resistance – RCSC Specification for
Structural Joints Using ASTM A325 and A490 Bolts**

Example: 2L 5 x 3 1/4
 $P_u = 46.8k$ $F_y = 36 \text{ ksi}$ $L = 150 \text{ in.}$
 $A = 3.88 \text{ in.}^2$ $I_y = 5.53 \text{ in.}^4$ $\bar{x} = 0.648 \text{ in.}$
 Double Angles have 3/8 in. separation

$$Q = \frac{3.88}{2} \left(0.648 + \frac{0.375}{2} \right) = 1.62 \text{ in.}^3$$

$$\text{S.F.} = 0.008(46.8)(150) \frac{1.62}{5.53} = 16.45k$$

The shear force (single shear) is 35% of P_u
 For a slip = 1/32 in. at the ends,
 the mid-height deflection will be 2 1/4 in.



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SUMMARY

- A better understanding of the stability design for various structural shapes has been provided.
- The design limit states of flexural, torsional, and flexural-torsional buckling resistance of compression members have been reviewed.
- The most frequently applied AISC LRFD and ASD design strengths for compression members provided in the Steel Specification have been discussed.
- The use of design examples and video demonstrations has helped our understanding of stability design.

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POLLING QUESTION 2

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Up Next...

Session 3: June 26 –
Behavior of Flexural Members
by R.D. Ziemian, PE, PhD

Using an approach similar to that employed in Session 1, this lecture will begin by presenting and dissecting the solution to the differential equation that defines the elastic lateral torsional buckling (LTB) strength of beams. Related flexural and torsional concepts, including the benefits of warping resistance, will be briefly reviewed. The assumption of elastic behavior will then be relaxed to define the inelastic LTB and plastic moment capacities of flexural members. The strength of beams without slender elements will be covered and ultimately presented in the form of beam resistance curves.

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- REINFORCEMENT – Reinforce what you learned tonight. Get more out of the course.

NOTE: If you attend the live presentation, you do not have to take the quizzes to receive CEUs/PDHS.



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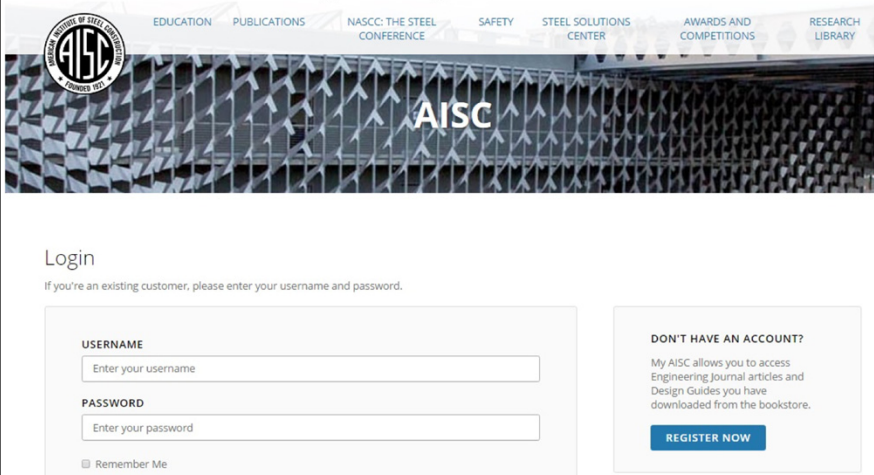
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Event	Start Date
NS 13 8-Session Package-Night School 13 - Design of Industrial Buildings	1/30/2017 7:00:00 PM
NS 14 8-Session Package-Night School 14 - Fundamentals of Stability	6/5/2017 7:00:00 PM

Night School Resources for 8-session package Registrants




Night School 13: Design of Industrial Buildings

8-SESSION PACKAGE RESOURCES

Event	Date	Handouts	Video	Quiz	Attendance
NS13 - Design Criteria	1/30/2017 7:00:00 PM	Handouts	View Passcode: NS13DSN	Pass Score: 80	Pending
NS13 - Economic Considerations	2/6/2017 7:00:00 PM	Handouts	Available 02/08/2017 5pm EST	Available 02/08/2017 5pm EST	Pending
NS13 - Lateral Load Systems and Details	2/13/2017 7:00:00 PM	Handouts	Available 02/15/2017 5pm EST	Available 02/15/2017 5pm EST	Pending
NS13 - Preliminary Design Procedures	2/27/2017 7:00:00 PM	Handouts	Available 03/02/2017 5pm EST	Available 03/02/2017 5pm EST	Pending
NS13 - Crane Girder Design and Frame Analysis	3/6/2017 7:00:00 PM	Handouts	Available 03/08/2017 5pm EST	Available 03/08/2017 5pm EST	Pending
NS13 - Frame Member and Connection Design	3/13/2017 7:00:00 PM	Handouts	Available 03/15/2017 5pm EST	Available 03/15/2017 5pm EST	Pending
NS13 - Transfer Crane Girder & Longitudinal Bldg Bracing Dn	3/27/2017 7:00:00 PM	Handouts	Available 03/29/2017 5pm EST	Available 03/29/2017 5pm EST	Pending
NS13 - Building Envelope and Bracing Design	4/3/2017 7:00:00 PM	Handouts	Available 04/05/2017 5pm EST	Available 04/05/2017 5pm EST	Pending
NS13 - Final Exam	4/10/2017 7:00:00 PM			Available 04/12/2017 5pm EST	

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- Weekly “quiz and recording” email.
- Weekly updates of the master Quiz and Attendance record found at www.aisc.org/nightschool. Scroll down to Quiz and Attendance records.
 - Updated on Tuesday mornings.



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- Webinar connection information:
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- Link to handouts also found here.

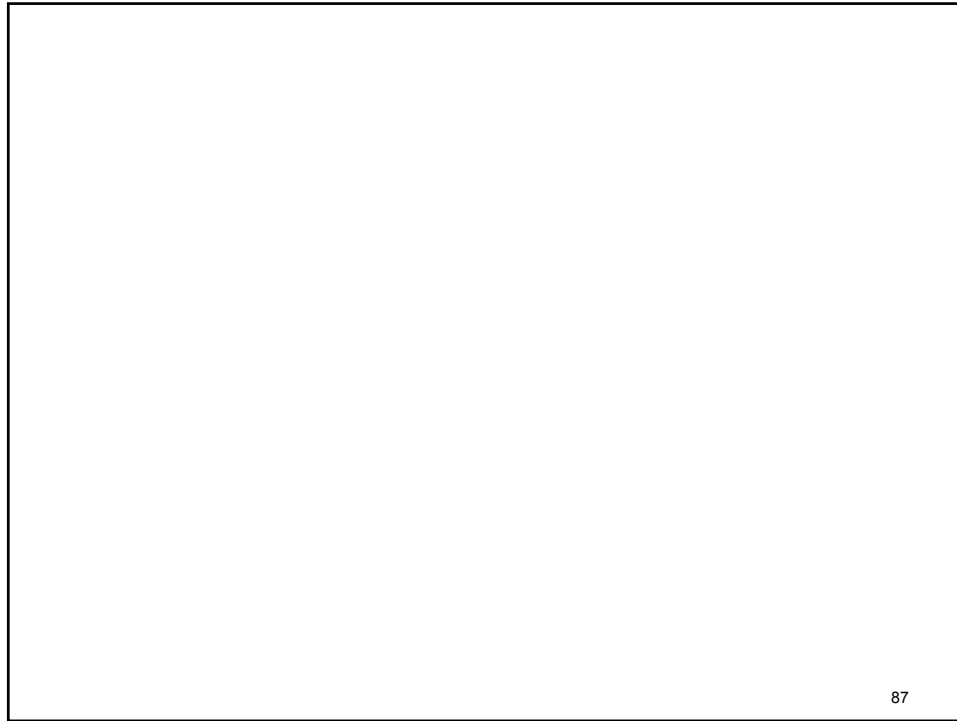


Thank You

Please give us your feedback!
Survey at conclusion of webinar.

There's always a solution in steel.





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**TORSIONAL BUCKLING
 CRUCIFORM MEMBERS**

$$\bar{r}_o^2 = x_o^2 + y_o^2 + \frac{I_x + I_y}{A_g} \quad (E4 - 11)$$

$$F_{crz} = \frac{GJ}{A_g \bar{r}_o^2} \quad (E4 - 3)$$

88

**TORSIONAL BUCKLING
 CRUCIFORM MEMBERS**

Twist about the shear center of the section.

89

**TORSIONAL BUCKLING
 CRUCIFORM MEMBERS**

$$J = \frac{4bt^3}{3}$$

$$I_x = I_y = 2 \frac{b^3t}{3}$$

$$F_{crz} = \frac{GJ}{A\bar{r}_c^2} = \frac{GJ}{I_x + I_y} = \frac{G}{\left(\frac{b}{t}\right)^2} = \frac{0.385E}{\left(\frac{b}{t}\right)^2}$$

90

DESIGN STRENGTH
SINGLE ANGLE COLUMN L 2 x 2 x 3/16

Buckling about z-axis

$$\left(\frac{KL}{r}\right)_z = \frac{44}{0.389} = 113 \leq 4.71 \sqrt{\frac{E}{F_y}} = 133 \quad F_y = 36 \text{ ksi}$$

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = 22.41 \text{ ksi}$$

$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}}\right] F_y = 18.38 \text{ ksi}$$

$$P_n = F_{cr} A_g = 13.27 \text{ ksi}$$



$$\text{ASD } \frac{P_n}{\Omega} = 7.95 \text{ kips} \quad \text{LRFD } \phi P_n = 11.94 \text{ kips}$$

Buckling about z-axis does not control

91

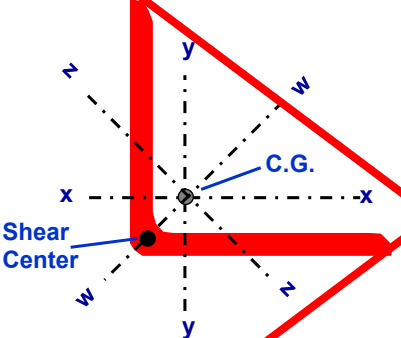
COMPRESSION MEMBER
DESIGN EXAMPLE 4

*ANSI/AISC 360-10 Specification
 for the Design of Steel Buildings, 2010*

92

MEMBER DESIGN STRENGTH
SINGLE ANGLE COLUMN $\angle 6 \times 6 \times 5/8$



Section Properties:

$A_g = 7.13 \text{ in.}^2$
 $I_x = I_y = 24.1 \text{ in.}^4$
 $J = 0.955 \text{ in.}^4$
 $C_w = 2.5 \text{ in.}^6$
 $r_x = 1.84 \text{ in.}$
 $r_z = 1.17 \text{ in.}$
 $r_o = 3.28 \text{ in.}$
 $Q = 1.0$
 $H = 0.631$

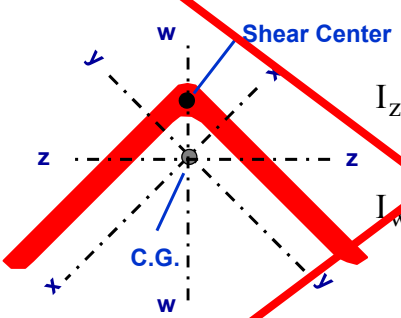
Material Properties:

$F_y = 50 \text{ ksi}$
 $E = 29,000 \text{ ksi}$
 $G = 0.385 E$

Length: $KL = L = 10 \text{ ft.}$

93

MEMBER DESIGN STRENGTH
SINGLE ANGLE COLUMN $\angle 6 \times 6 \times 5/8$
Flexural-Torsional Buckling



$I_z = A_g r_z^2 = (7.13)(1.17)^2 = 9.76 \text{ in.}^2$
 $I_w = I_x + I_y - I_z$
 $= 2(24.1) - 9.76 = 38.44 \text{ in.}^4$
 $r_w = \sqrt{\frac{I_w}{A_g}} = \sqrt{\frac{38.44}{7.13}} = 2.32 \text{ in.}$

94

MEMBER DESIGN STRENGTH
SINGLE ANGLE COLUMN L 6 x 6 x 5/8
Flexural-Torsional Buckling

$$F_{ez} = \left(\frac{\pi^2 E C_w}{(K_z L)^2} + GJ \right) \frac{1}{A_g \bar{r}_o^2} \approx \frac{GJ}{A_g \bar{r}_o^2}$$

↑

Small compared to GJ, therefore this term can be safely ignored for angles, double angles and tees
In this case, 50 versus 10700

$$F_{ez} = \frac{GJ}{A_g \bar{r}_o^2} = \frac{(11200)(0.955)}{(7.13)(3.28)^2} = 139.44 \text{ ksi}$$

95

MEMBER DESIGN STRENGTH
SINGLE ANGLE COLUMN L 6 x 6 x 5/8
Flexural-Torsional Buckling

$$F_{ew} = \frac{\pi^2 E}{\left(\frac{KL}{r} \right)_w^2} = 106.98 \text{ ksi}$$

$$F_e = \frac{F_{ew} + F_{ez}}{2H} \left[1 - \sqrt{1 - \frac{4F_{ew} F_{ez} H}{(F_{ew} + F_{ez})^2}} \right] \quad (E4 - 5)$$

$$F_e = \frac{106.98 + 139.44}{2(0.631)} \left[1 - \sqrt{1 - \frac{4(106.98)(139.44)(0.631)}{(106.98 + 139.44)^2}} \right]$$

$$F_e = 74.90 \text{ ksi}$$

96

DESIGN STRENGTH
SINGLE ANGLE COLUMN L 6 x 6 x 5/8
Buckling about z-axis

$$\left(\frac{KL}{r}\right)_z = \frac{(12)(10)}{1.17} = 103 \leq 4.71 \sqrt{\frac{E}{F_y}} = 113 \quad \text{OK}$$

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)_z^2} = 27.21 \text{ ksi} \qquad \frac{F_y}{F_{ez}} = \frac{50}{27.21} = 1.837$$

$$F_{cr} = \left[0.658 \frac{F_y}{F_e} \right] F_y = 23.18 \text{ ksi} \qquad \text{(E3 - 2)}$$

$$P_n = F_{cr} A_g = 165.25 \text{ kips} \qquad \text{(E3 - 1)}$$

➔ ASD $\frac{P_n}{\Omega} = 98.95 \text{ kips}$ LRFD $\phi P_n = 148.72 \text{ kips}$

97

MEMBER DESIGN STRENGTH
SINGLE ANGLE COLUMN L 6 x 6 x 5/8
Flexural-Torsional Buckling

$$\left(\frac{KL}{r}\right)_{F-T} = \frac{(12)(10)}{2.32} = 52 \leq 4.71 \sqrt{\frac{E}{F_y}} = 113 \quad \text{OK}$$

$$\frac{F_y}{F_{eF-T}} = \frac{50}{74.90} = 0.667$$

$$F_{crF-T} = \left[0.658 \frac{F_y}{F_e} \right] F_y = 37.82 \text{ ksi}$$

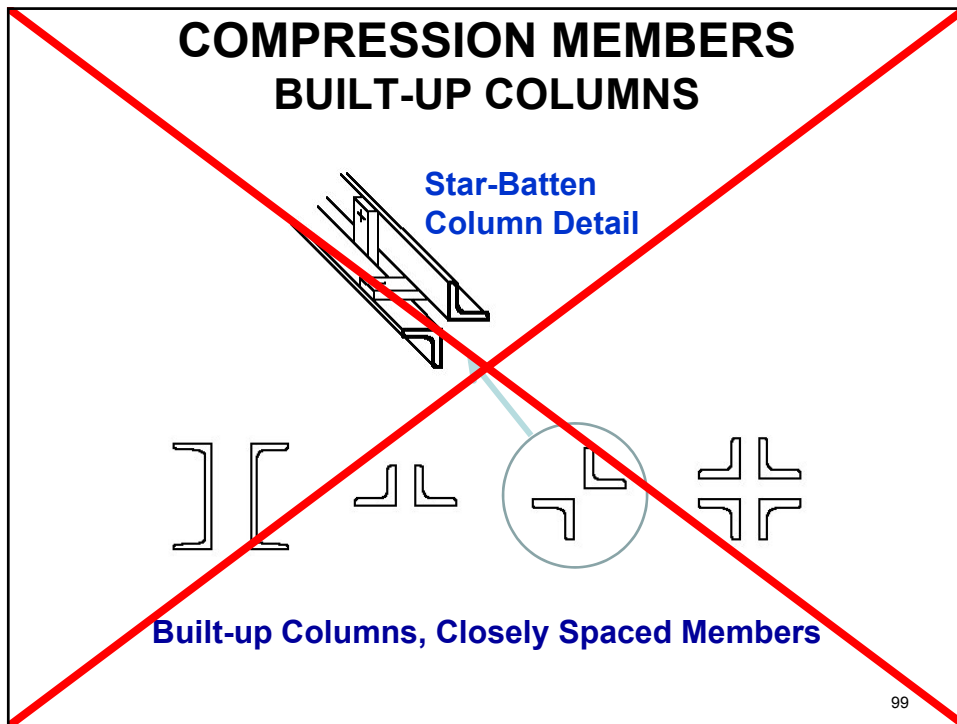
$$P_n = F_{cr} A_g = 269.66 \text{ kips}$$

ASD $\frac{P_n}{\Omega} = 161.47 \text{ kips}$ LRFD $\phi P_n = 242.69 \text{ kips}$

Buckling about z-axis controls

98





POLLING QUESTION 1

The AISC 360-10 Specification defines which buckling limit states for compression members?

- A. Flexural and Local Plate
- B. Torsional and Flexural-Torsional
- C. Flexural and Distortional

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