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## Course Description

### **Session 6: Building Analysis and Diaphragm Design**

**March 26, 2018**

This session will review various analysis types and applicability to seismic design. The session will address effective structural modeling, including moment releases and effective stiffness. This session will also discuss second-order effects in the analysis, and calculating drift. The session will also address diaphragm design including determination of building-analysis forces, capacity-design forces and design of members at diaphragm openings.





## Learning Objectives

- Identify various analysis types and their applicability in seismic design.
- Identify how to properly model moment releases and effective stiffness.
- Describe the attributes of P- $\delta$  and P- $\Delta$  effects.
- Describe the components of designing members at diaphragm openings.



There's always a solution in steel.

## Seismic Design in Steel: Concepts and Examples

Session 6: Building Analysis and Diaphragm Design  
March 26, 2018



Rafael Sabelli, SE



## Course objectives

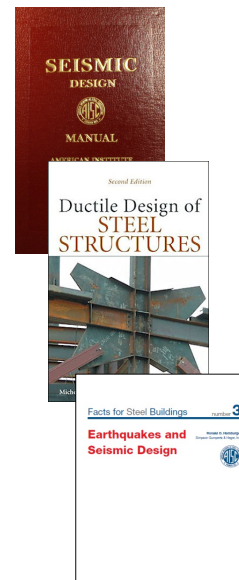
- Understand the principles of seismic design of steel structures.
- Understand the application of those principles to two common systems:
  - Special Moment Frames
  - Buckling-Restrained Braced Frames.
- Understand the application of design requirements for those systems.



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## Resources

- *AISC Seismic Design Manual*
- *Ductile Design of Steel Structures*, Bruneau, Uang, and Sabelli, McGraw Hill.
- *Earthquakes and Seismic Design*, Facts for Steel Buildings #3. Ronald O. Hamburger, AISC.
- Other publications suggested in each session



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## Other resources

- AISC Solutions Center
  - 866.ASK.AISC (866-275-2472)
  - Solutions@AISC.org
- AISC Night School
  - Nightschool@AISC.org



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## Course outline

### **Part I: Concepts**

1. Introduction to effective seismic design
2. Seismic design of moment frames
3. Seismic design of braced frames
4. Seismic design of buildings



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## Course outline

### **Part II: Application**

5.Planning the seismic design

### **6. Building analysis and diaphragm design**

7.Design of the moment frames

8.Design of the braced frames



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## **Session 6: Building analysis and Diaphragm design**



## Session topics

- Building Analysis
  - Lateral analysis methods
  - Load cases
  - Structural model
  - Design for stability & 2<sup>nd</sup>-order analysis
  - Strength-design forces
- Diaphragm Design
  - Diaphragm forces
  - Capacity-design forces
  - Diaphragm analysis
  - Collector design
  - Collector-connection design
  - Diaphragm openings



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## Building Analysis: Lateral analysis methods



## Lateral analysis methods

- Equivalent Lateral Force (ELF)
  - Standard procedure
  - Slightly conservative
- Modal Response Spectrum Analysis (MRSA)
  - ASCE 7-10: Scaled to 85% of ELF base shear
  - **ASCE 7-16:** Scaled to 100% of ELF base shear
  - Always:  $(M_{ot}/V)_{MRSA} < (M_{ot}/V)_{ELF}$
  - Slightly cumbersome
- Response History Analysis
  - Cumbersome; only used for special conditions



ASCE 7 §12.9.4.1

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## Lateral analysis methods

- 2D Analysis
  - 2D with flexible diaphragms
  - 2D with rigid or semi-rigid diaphragms
    - No torsional irregularity
    - Parallel and orthogonal frames (no skewed frames)
    - No out-of-plane offsets of the seismic system
- 3D
  - Everything else
  - Technically required in some cases for flexible diaphragms, but results are the same as 2D
    - Orthogonal combination requirements are not waived



ASCE 7 §12.7.3

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## Lateral analysis methods

- This example
  - Diaphragms are rigid per ASCE 7 §12.3.1.2
    - Concrete-filled steel deck
    - Span-to-depth ratio  $< 3$
    - No Horizontal irregularities
  - Use ELF for simplicity & clarity
    - MRSA slightly more economical
  - Use 2D analysis for simplicity & clarity
    - 3D slightly more economical



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## Load cases

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## Load Cases

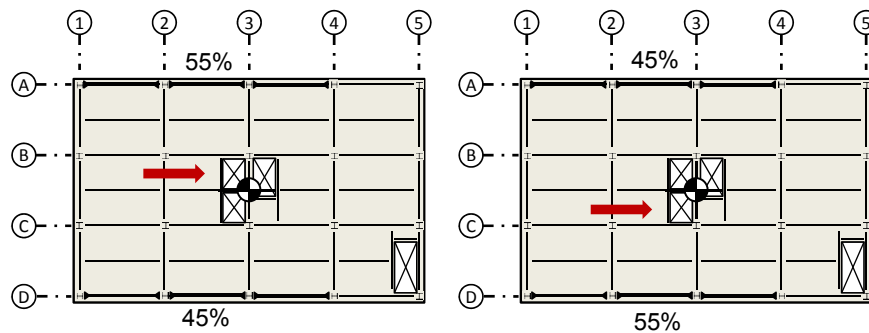
- 2 principal axes
  - Orthogonal combination not required
    - No skewed frames in this example
    - No columns shared by orthogonal frames
- Accidental torsion
  - Required for non-flexible diaphragms
  - 5% offset of mass in either direction
    - For simplicity, conservatively neglect torsional resistance of orthogonal frame (that is, 2D analysis)



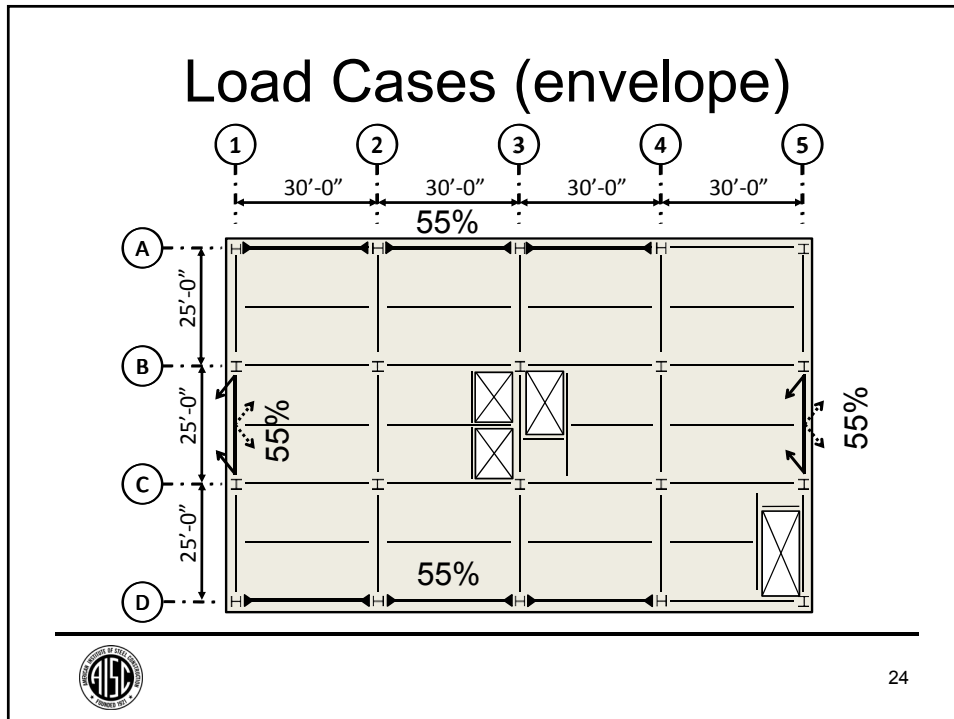
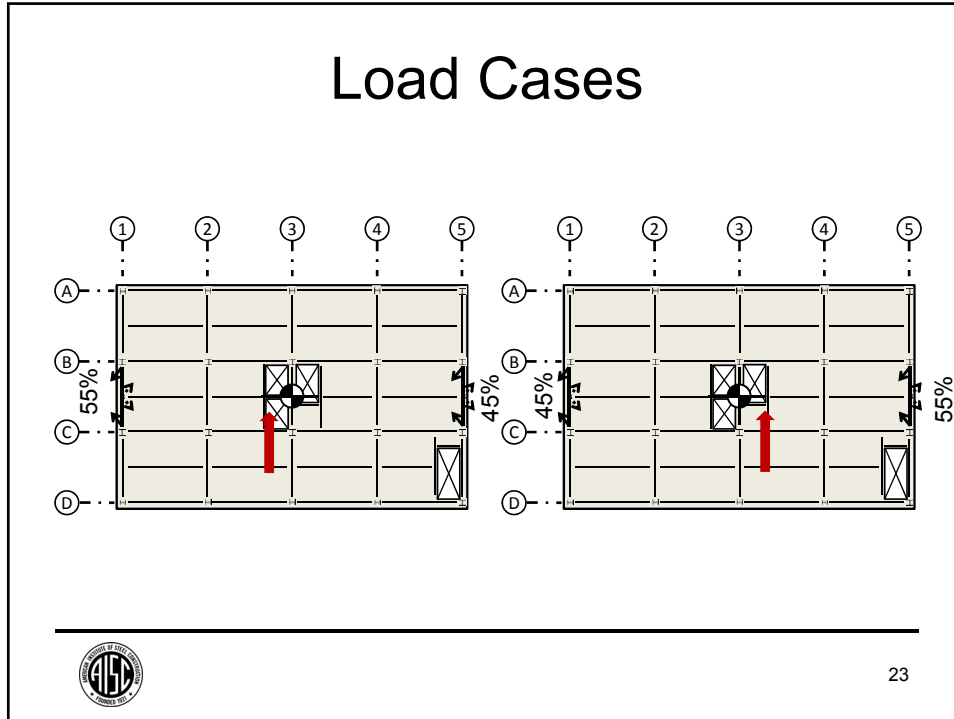
ASCE 7 §12.8.4.2

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## Load Cases



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## Base Shear

- From Session 5
  - SMF
    - $V = 0.0818 * 3313K$
    - $= 271K$
  - BRBF
    - $V = 0.100 * 3313K$
    - $= 331K$

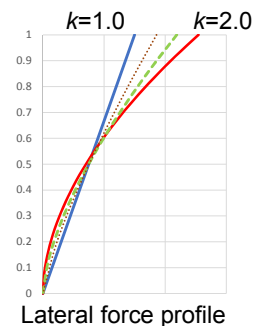


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## Vertical Distribution

$$C_{vx} = \frac{w_x h_x^k}{\sum_{i=1}^n w_i h_i^k}$$

- $T \leq 0.5s$        $k=1.0$
- $T \geq 2.5s$        $k=2.0$
- $0.5s < T < 2.5s$  interpolate



ASCE 7 §12.8.3

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## Vertical Distribution

$$C_{vx} = \frac{w_x h_x^k}{\sum_{i=1}^n w_i h_i^k}$$

- SMF
  - $C_u T_a = 0.918s$
  - $k=1.21$

Level	$w_i$ , kips	$h_i$ , (ft)	$w_i h_i^k$	$C_{vx}$	$F_x$ , (kips)
Roof	708.4	51.5	83,115	0.373	101
4 <sup>th</sup>	868.1	39	72,778	0.327	89
3 <sup>rd</sup>	868.1	26.5	45,617	0.205	55
2 <sup>nd</sup>	868.1	14	21,092	0.095	26
Total	3312.9		222,602	1.000	271



ASCE 7 §12.8.3

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## Vertical Distribution

$$C_{vx} = \frac{w_x h_x^k}{\sum_{i=1}^n w_i h_i^k}$$

- BRBF
  - $C_u T_a = 0.100s$
  - $k=1.125$

Level	$w_i$ , kips	$h_i$ , (ft)	$w_i h_i^k$	$C_{vx}$	$F_x$ , (kips)
Roof	708	51.5	59,687	0.362	120
4 <sup>th</sup>	868	39.0	53,499	0.325	108
3 <sup>rd</sup>	868	26.5	34,639	0.210	70
2 <sup>nd</sup>	868	14.0	16,898	0.103	34
Total	3313		164,723	1.000	331



ASCE 7 §12.8.3

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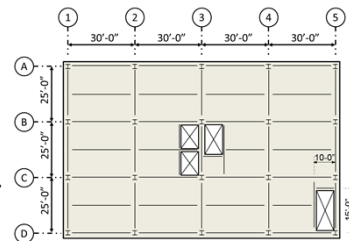
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## Structural model



## Structural Model

- Diaphragms are
  - Flexible,
  - Rigid, or
  - Semi-rigid
- Rigid diaphragms assumed, if
  - Concrete deck
  - $L/d \leq 3$ 
    - $120/75 = 1.6$  OK
  - No horizontal irregularities



ASCE 7 §12.3.1

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## Structural Model

- Gravity framing (if included in model)
  - Design seismic system for 100% of lateral forces
  - Prevent shear in gravity columns in seismic analysis
    - Pin columns top and bottom
      - Except SMF
    - Pin non-frame beams connecting to SMF columns
- Column size change at floor



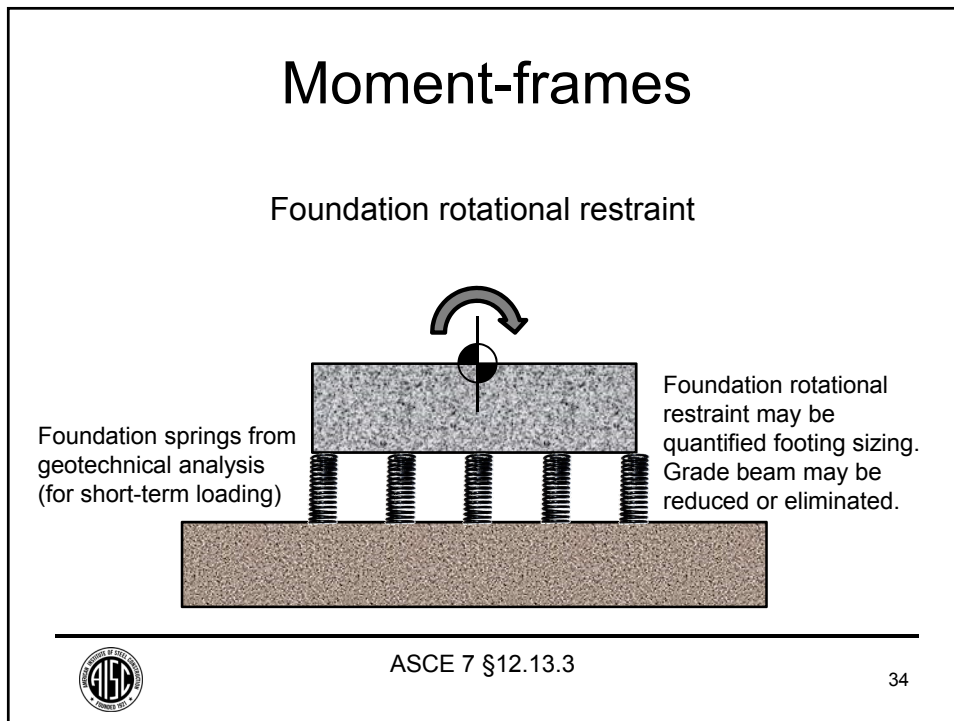
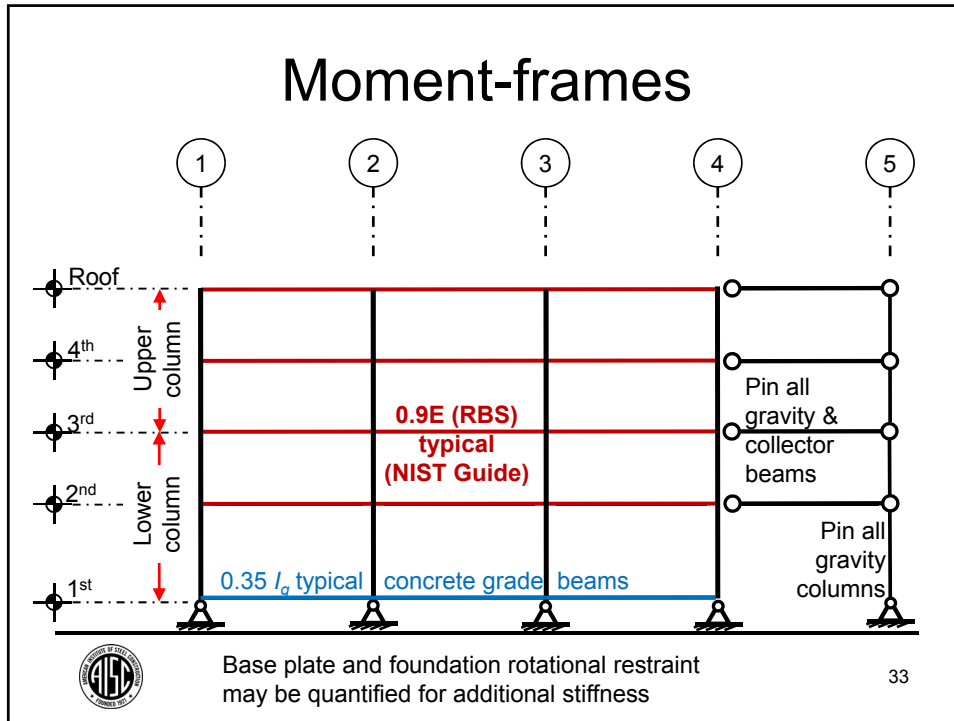
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## Moment frames

- RBS connection
  - Reduced beam stiffness
    - (90% if maximum RBS reduction)
  - Prequalification limits on members
- No rigid-end offset
- Mesh columns into 4 segments (for  $P-\delta$  effects, dependent on software capability)
  - Or use  $B_1$  factor
- Use non-composite beam stiffness
- Do not assume fully rigid bases



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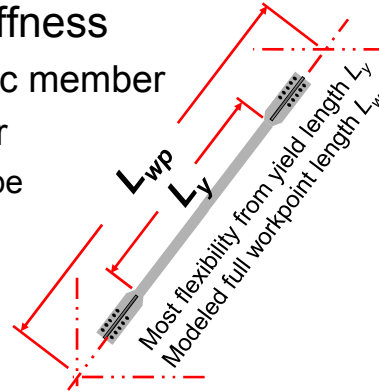


## Braced frames

- Pin all members
- Use increased brace stiffness
  - Represents non-prismatic member
  - $K_F = 1.4$  per manufacturer
    - Varies with connection type
    - Varies with area
    - Varies with length

$$K \sim EA_{sc} / L_y = K_F EA_{sc} / L_{wp}$$

$$1.2 \leq K_F \leq 1.8$$



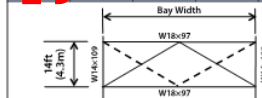
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### APPROXIMATE STIFFNESS MODIFICATION FACTORS, $K_F^{1,2,7}$

Sizes shown are representative of typical BRB sizes. Information on intermediate and larger sizes is available upon request.

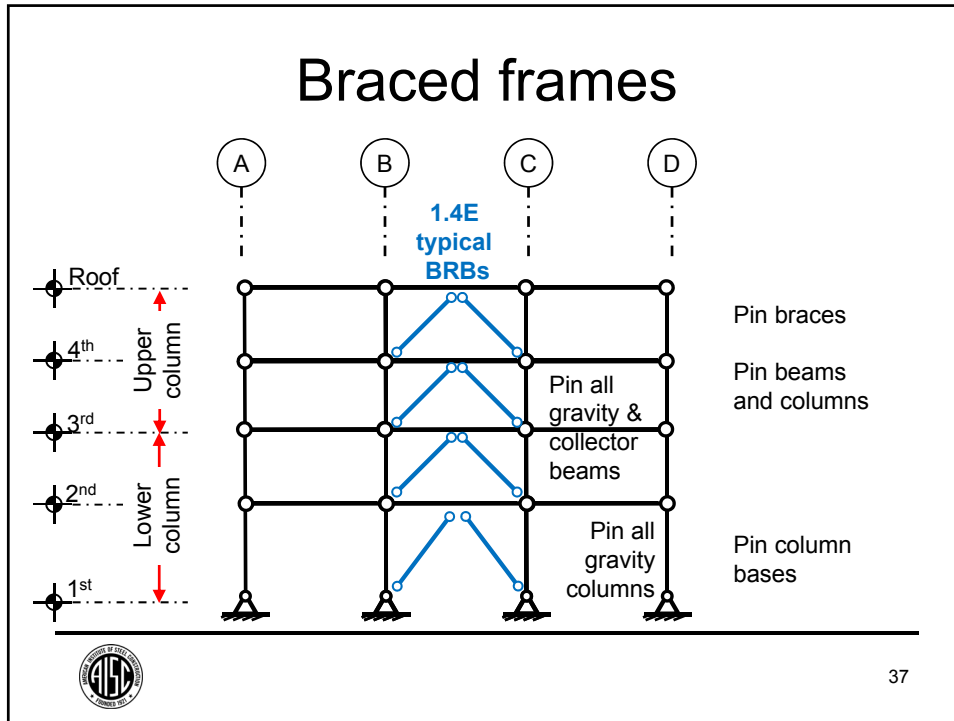
$F_y = 38$ ksi (262 MPa)	$A_g$ , in <sup>2</sup> (cm <sup>2</sup> )	$P_y$ , axial, kip (kN)	Bay Width, ft (m)									
			15 (4.6)	20 (6.1)	25 (7.6)	30 (9.1)	35 (10.7)	30 (9.1)	35 (10.7)	40 (12.2)	45 (13.7)	50 (15.2)
			SINGLE DIAGONAL					CHEVRON/V				
2.0 (13)	68 (306)		1.37	1.34	1.31	1.29	1.28	1.33	1.30	1.27	1.25	1.24
3.0 (19)	103 (448)		1.36	1.33	1.31	1.29	1.27	1.32	1.29	1.27	1.25	1.23
4.0 (26)	137 (613)		1.47	1.42	1.37	1.35	1.32	1.42	1.37	1.34	1.31	1.29
5.0 (32)	171 (754)		1.41	1.37	1.34	1.31	1.29	1.36	1.33	1.30	1.28	1.26
6.0 (39)	205 (919)		1.51	1.45	1.40	1.37	1.34	1.45	1.41	1.37	1.34	1.31
8.0 (52)	274 (1225)		1.53	1.47	1.42	1.38	1.36	1.47	1.42	1.38	1.35	1.33
9.0 (58)	308 (1367)		1.54	1.47	1.42	1.39	1.36	1.48	1.43	1.39	1.36	1.33
10.0 (65)	342 (1532)		1.59	1.51	1.45	1.41	1.38	1.52	1.47	1.42	1.38	1.35
11.0 (71)	376 (1673)		1.60	1.51	1.46	1.41	1.38	1.53	1.47	1.43	1.39	1.36
12.0 (77)	410 (1814)		1.67	1.57	1.50	1.45	1.41	1.59	1.53	1.47	1.43	1.39
14.0 (90)	479 (2121)		1.60	1.52	1.46	1.42	1.39	1.53	1.47	1.43	1.40	1.37
16.0 (103)	547 (2427)		1.69	1.59	1.52	1.47	1.43	1.61	1.54	1.49	1.44	1.41
18.0 (116)	616 (2733)		1.69	1.59	1.52	1.47	1.43	1.61	1.54	1.49	1.45	1.41
20.0 (129)	684 (3040)		1.66	1.57	1.50	1.46	1.42	1.59	1.52	1.47	1.44	1.40
22.0 (142)	752 (3346)		1.76	1.65	1.57	1.51	1.47	1.68	1.60	1.53	1.48	1.44
24.0 (155)	821 (3652)		1.81	1.68	1.59	1.53	1.47	1.69	1.61	1.57	1.52	1.48
26.0 (168)	889 (3959)		1.82	1.69	1.60	1.54	1.46	1.66	1.58	1.59	1.53	1.48
28.0 (181)	958 (4265)		1.83	1.70	1.61	1.55	1.48	1.69	1.61	1.59	1.54	1.49
30.0 (194)	1026 (4571)		1.81	1.69	1.60	1.54	1.49	1.70	1.62	1.58	1.53	1.48
Workpoint Length, ft (m)			20.5 (6.3)	24.4 (7.4)	28.7 (8.7)	33.1 (10.0)	37.7 (11.5)	20.5 (6.3)	22.4 (6.8)	24.4 (7.4)	26.5 (8.1)	28.7 (8.7)

STORY HEIGHT: 14ft (4.3m)



Brochure from Core Brace

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## Design for stability & Second-order analysis

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## Design for stability AISC 360

- Direct Analysis Method (DAM)
  - Analysis
    - Decreased member stiffness
      - Different model used for drift, period
    - 2<sup>nd</sup>-order effects must be included
  - Can be used for any level of second-order effect
    - But  $B_2 > 1.5$  is unusual
  - $K=1$  for columns
  - Requires minimum lateral load



AISC 360 §C1.1

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## Design for stability AISC 360

- Effective Length Method (ELM)
  - Analysis
    - 2<sup>nd</sup>-order effects must be included
    - Uses same model for strength, drift, period
  - Reduced column strength
  - Can be used for  $B_2 \leq 1.5$
  - $K=1$  for
    - Braced-frame columns
    - Moment-frame columns with  $B_2 \leq 1.1$
  - Requires minimum lateral load



AISC 360 Appendix 7 §7.2

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## Design for stability AISC 360

- First-Order Analysis Method (FOAM)
  - Analysis
    - 2<sup>nd</sup>-order effects not included
    - Uses same model for strength, drift, period
  - 2<sup>nd</sup>-order effects addressed by additional lateral load
    - $N_i = 2.1 \frac{\Delta}{h} Y_i \approx 0.008 Y_i$  for SMF at drift limit of  $0.02h$
    - Increase  $C_s$  by 12% in this example ( $0.0818 + 0.0099 = 0.0917$ )
      - Not a penalty if the frame is governed by drift
  - Can be used for  $B_2 \leq 1.5$
  - $K=1$  for columns
  - $P_u/P_v$  for columns  **$\leq 0.5$**



AISC 360 Appendix 7 §7.3

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## Design for stability: Braced Frames

- Typically governed by strength
- Use ELM (Appendix 7 §7.2)
  - Simpler to use same model for drift
- Use  $K=1$ 
  - Always OK for braced frames
- Perform 2<sup>nd</sup> order analysis
  - Or use  $B_2$



AISC 360 Appendix 7

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## Design for stability: Moment Frames

- Typically governed by drift
- Use ELM if  $B_2 \leq 1.1$
- If  $B_2 > 1.1$ 
  - Size frame for drift
  - Check strength with 1<sup>st</sup>-order analysis (Appendix 7 §7.3)
    - Supplemental lateral load  $\sim 0.084 P_{story}$  for SMF at drift limit
      - $B_2$  not required
      - Or DAM
  - Don't try to calculate  $K$  factors!
- If  $B_2 > 1.5$  redesign!
  - Will not meet ASCE 7 §12.8.7 stability check



AISC 360 Appendix 7

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## Second-order effects

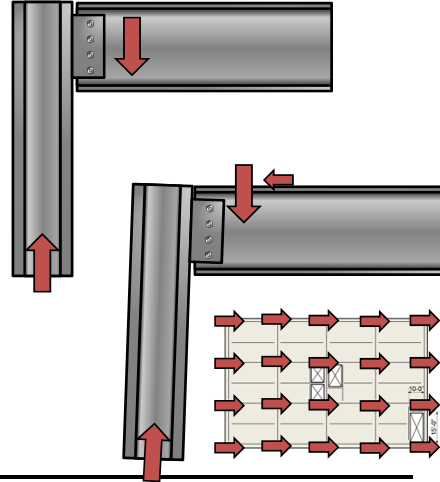
- Second-order analysis
  - Equilibrium in deformed condition
  - Analysis must include all gravity load
- Approximate second-order analysis
  - Appendix 8
  - $B_1$  Beam-columns
  - $B_2$  The entire lateral-load-resisting system
- ASCE 7 §12.8.7 requires consideration of second order effects for forces and displacements when  $\theta > 0.1$  ( $\theta \sim 1 - 1/B_2 < B_2 - 1$ )



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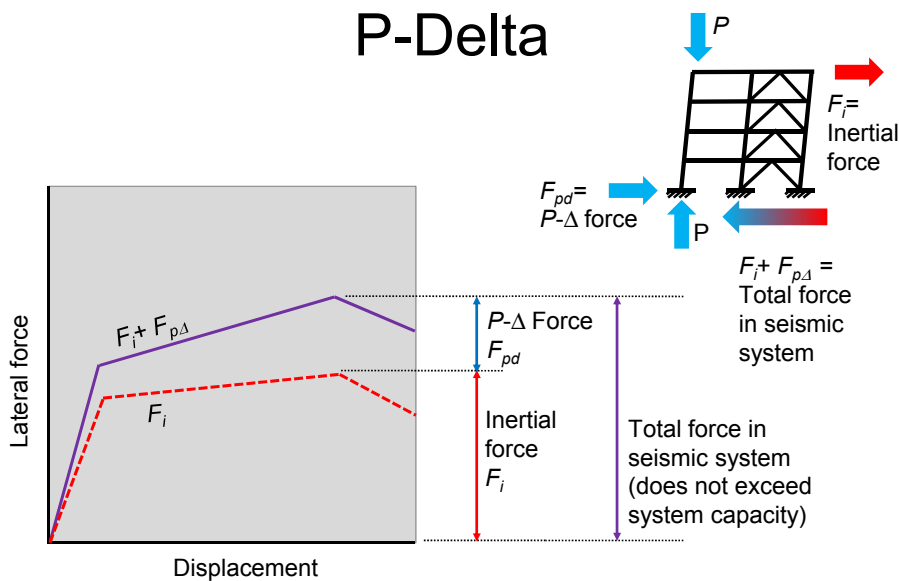
## Second-order effects (P-Delta)

- Columns resist the story gravity force
- Lateral loads induce drift
  - Columns slope
  - Column axial force has horizontal component
  - Horizontal component is additional thrust on diaphragm



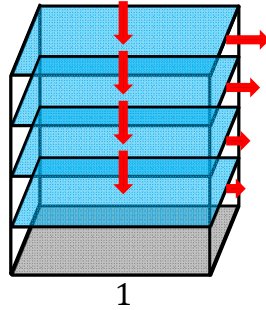
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## P-Delta



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## Second-order effects (approximate 2<sup>nd</sup>-order analysis)



- $B_2$  calculation
  - $P_{story}$ 
    - Gravity load acting on story
    - Includes gravity load above
    - Use consistent dead load with seismic mass
    - Use live load from load combination
  - $V_{story}$ 
    - Story shear

$$B_2 = \frac{1}{1 - \frac{P_{Story}\Delta_{elastic}}{R_m V_{Story} h}}$$

Largest  $B_2$  at base for seismic



AISC 360 Appendix 8

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## Second Order Effects: SMF

$$B_2 = \frac{1}{1 - \frac{P_{Story}\Delta_{elastic}}{\left[1 - 0.15 \frac{P_{mf}}{P_{Story}}\right] V_{Story} h}}$$

$$B_2 = \frac{1}{1 - \frac{0.0036 P_{Story}}{0.93 V_{Story}}}$$

	$B_2$
Roof	1.02
4 <sup>th</sup>	1.03
3 <sup>rd</sup>	1.03
2 <sup>nd</sup>	1.04

- Assume drift-controlled
  - $\Delta_{elastic} = 0.02h/C_d$   
 $= 0.0036h$
- Use  $P-\Delta$  gravity load
  - $P = 1.0D + 0.5L$
- Assume  $P_{mf} = \frac{1}{2} P_{story}$
- $B_2 = 1.04$  (largest)  $< 1.1$ 
  - Use ELM
  - $K=1$



AISC 360 Appendix 8

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## Second Order Effects: BRBF

$$B_2 = \frac{1}{1 - \frac{P_{Story} \Delta_{elastic}}{V_{Story} h}}$$

$$B_2 = \frac{1}{1 - \frac{0.004 P_{Story}}{V_{Story}}}$$

	$B_2$
Roof	1.02
4 <sup>th</sup>	1.03
3 <sup>rd</sup>	1.03
2 <sup>nd</sup>	1.04

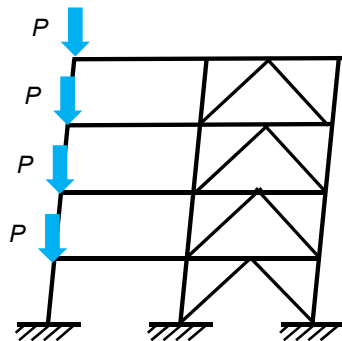
- Assume drift
  - $\Delta_{elastic} = 0.02h/C_d$   
 $= 0.004h$
- Use  $P-\Delta$  gravity load
  - $P=1.0D+0.5L$
- $B_2 = 1.04$  (largest)



AISC 360 Appendix 8

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## Second Order Effects: Analysis



- Use “leaner” column to resist all gravity-column axial forces
- Use software capable of performing second-order analysis
- Results
  - Amplified lateral-system forces
  - Amplified drift



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## Second Order Effects

- ASCE 7 stability check

$$\theta = \frac{P_x \Delta I_e}{V_x h_{sx} C_d} = \frac{P_x \Delta_{elastic}}{V_x h_{sx}} = \frac{P_x / h_{sx}}{K_x} \quad \theta_{max} = \frac{0.5}{\beta C_d} \leq 0.25$$

- $\theta \sim 1 - 1/B_2 = 0.04$  (max)

$$\theta_{max} = \frac{0.5}{0.9 * 5} = 0.11$$

- OK

Where  $\beta$  is the ratio  
of shear demand to  
shear capacity



ASCE 7 §12.8.7

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## Stability & 2<sup>nd</sup>-order analysis

- Due to high seismic demands at this location, system is required to be stiff enough so that second-order effects are minor
  - First-order effects are large
    - Second-order effects are relatively small
  - At sites with low seismic demands second-order effects are more important
    - i.e.,  $B_2$  and  $\theta$  will be larger



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## Stability & 2<sup>nd</sup>-order analysis

- Use ELM for SMF & BRBF
  - Same model for strength and drift
    - No DAM stiffness reduction
    - No FOAMy supplemental lateral force
- For clarity in this example, second-order analysis software not used
  - Approximate second-order analysis (Appendix 8)
    - Amplify lateral forces and displacements by  $B_2$
    - Amplify non-sway (gravity) moments by  $B_1$

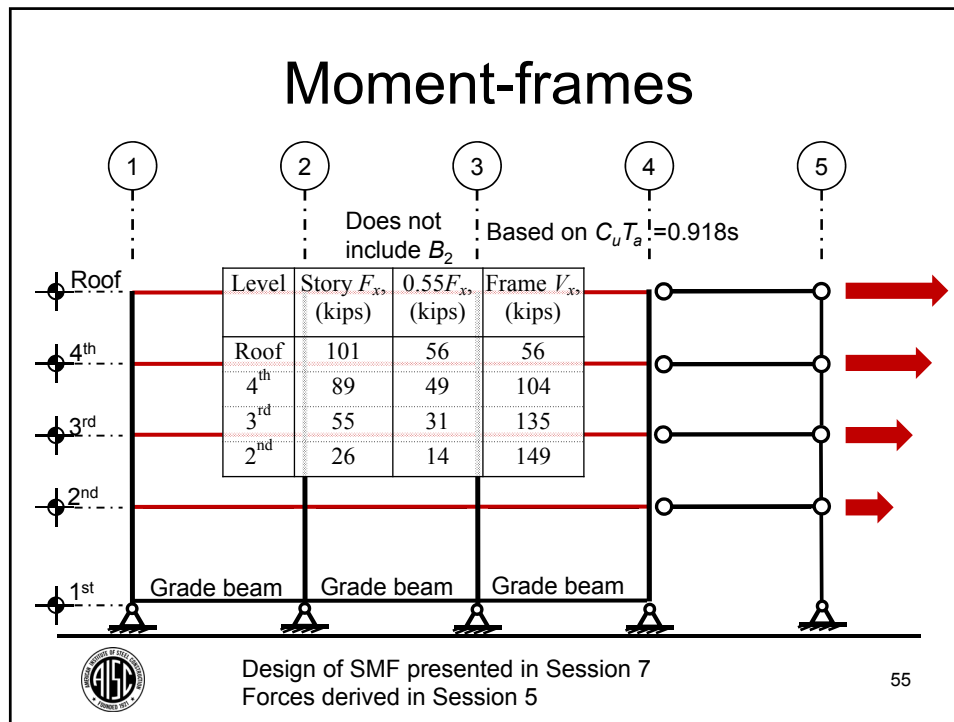


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## Analysis forces

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### Moment frames

- Moment frames likely drift-controlled
  - Design for drift
  - Check strength after member selection
- Design base shear strength check subject to maximum period  $C_u T_a$
- Drift not subject to maximum period
  - Design in Session 7 tracks period with iteration
  - Required stiffness (and thus period) can be approximated using spectrum & drift limit

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ASCE 7 §12.8.2.1

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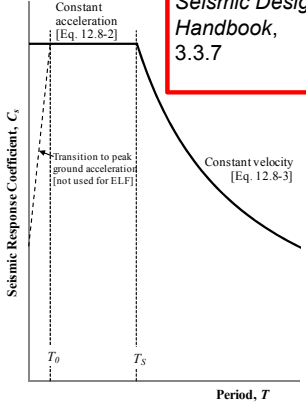


## Drift-determined period

Adapted from Naeim's *Seismic Design Handbook*, 3.3.7

$$S_a = C_s \left( \frac{R}{I_e} \right) g = \frac{S_{D1}g}{T/sec}$$

$$S_d = \frac{S_a}{\omega^2} = \frac{S_a T^2}{4\pi^2} = \frac{S_{D1}gT(sec)}{4\pi^2}$$



Seismic Response Coefficient,  $C_s$

Constant acceleration [Eq. 12.8-2]

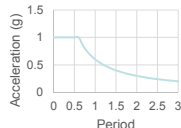
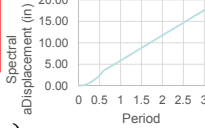
Transition to peak ground acceleration [not used for ELF]

Constant velocity [Eq. 12.8-3]

Period,  $T$


$T_0$        $T_s$

$S_d \approx \frac{2}{3} \Delta_{roof} = \frac{2}{3} 0.02h = \frac{S_{D1}gT(sec)}{4\pi^2}$ 
 $T = \frac{2}{3} \frac{0.02h4\pi^2}{S_{D1}g(sec)} = \frac{2}{3} \frac{0.025h(sec)}{S_{D1}(feet)}$ 
 $T = \frac{2}{3} \frac{0.025(51.5')(sec)}{0.6(feet)} = 1.4sec$


$\frac{2}{3}$  is an approximate correction factor

- Spectral displacement is not roof displacement
- 1<sup>st</sup>-mode mass participation <100%


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## Moment frames

- Design base shear based on maximum period  $C_u T_a = 0.92$  sec
- Drift-determined period = **1.4** sec
  - Corresponds to **2.1**  $T_a$
- Recommendation for SMF
  - For first iteration use either
    - Drift-determined period
    - $2.0 T_a$
  - Use calculated period for subsequent iteration


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### Braced frames

Level	Story $F_{x_s}$ (kips)	$0.55F_{x_s}$ (kips)	Frame $V_{x_s}$ (kips)
Roof	120	66	66
4 <sup>th</sup>	108	59	125
3 <sup>rd</sup>	70	38	164
2 <sup>nd</sup>	34	19	182

Does not include  $\rho=1.3$   
 Does not include  $B_2$

Design of BRBF presented in Session 8  
 Forces and redundancy from Session 5

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### Braced-frame forces

Level	Frame $F_{x_s}$ (kips)	$B_2$	$\rho$	Frame $\rho B_2 F_{x_s}$ (kips)	Frame $\rho B_2 V_{x_s}$ (kips)
Roof	66.0	1.02	1.30	87.6	88
4 <sup>th</sup>	59.2	1.03	1.30	79.3	167
3 <sup>rd</sup>	38.3	1.03	1.30	51.3	218
2 <sup>nd</sup>	18.7	1.04	1.30	25.3	243

Design of BRBF presented in Session 8  
 Forces derived in Session 5

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### Braced-frame forces

Level	Brace Force $\rho B_2 P_x$ (kips)
4 <sup>th</sup>	62
3 <sup>rd</sup>	118
2 <sup>nd</sup>	154
1 <sup>st</sup>	183

Assume braces resist 100% of story shear

$$P_u = \frac{F}{2 \cos \theta}$$

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## Diaphragm Design

There's always a solution in steel.

## Diaphragm design

- Typically done after design of frames
  - Requires consideration of transfer forces between frames
    - 3D building analysis
    - Indeterminate analysis using designed member stiffness
  - Forces may be limited by yielding elements
- This example
  - Done now to allow full sessions for each system



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## Diaphragm Forces

$$F_{px} = \frac{\sum_{i=x}^n F_i}{\sum_{i=x}^n W_i} w_{px} \quad 0.2S_{DS}Iw_{px} \geq F_{px} \geq 0.4S_{DS}Iw_{px}$$

Level	Story $F_{px}$ , (kips)	
Roof	141.7	$0.2S_{DS}Iw_{px}$
4 <sup>th</sup>	173.6	$0.2S_{DS}Iw_{px}$
3 <sup>rd</sup>	173.6	$0.2S_{DS}Iw_{px}$
2 <sup>nd</sup>	173.6	$0.2S_{DS}Iw_{px}$

$\rho=1.0$  for diaphragm design  
 (ASCE 7 §12.3.4.1)

$B_2$  (i.e., 2<sup>nd</sup> order amplification)  
 applies to diaphragm design  
 (ASCE 7 §12.8.7)



ASCE 7 §12.10.1

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There's always a solution in steel.

## Diaphragm analysis



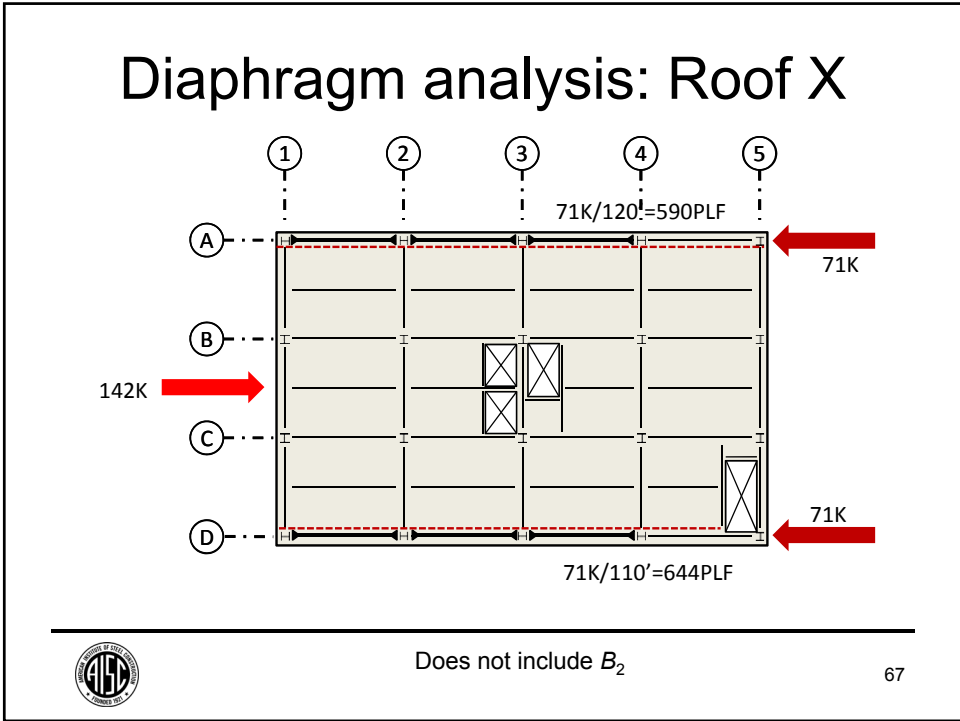
## Diaphragm analysis

- Determine diaphragm shear
- Determine collector forces
  - Apply  $\Omega_o$  factor per ASCE 7 §12.10.2
- Determine chord forces
  - Diaphragm equivalent-beam moments
  - Divide by depth
- $B_2$  applies to entire lateral analysis
  - Incorporated in member-design forces

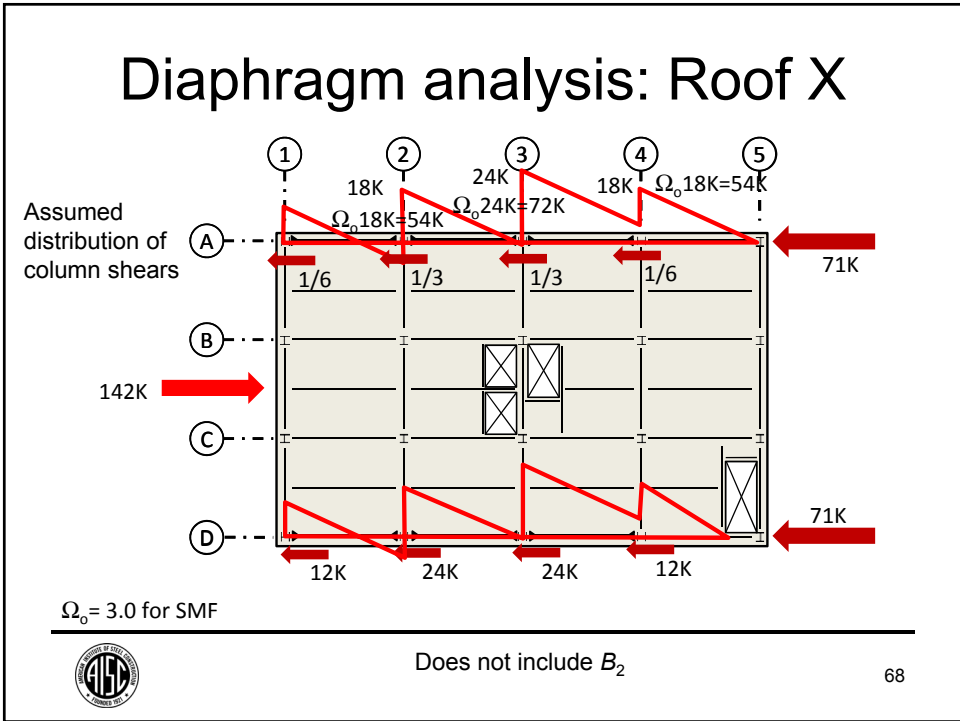


66

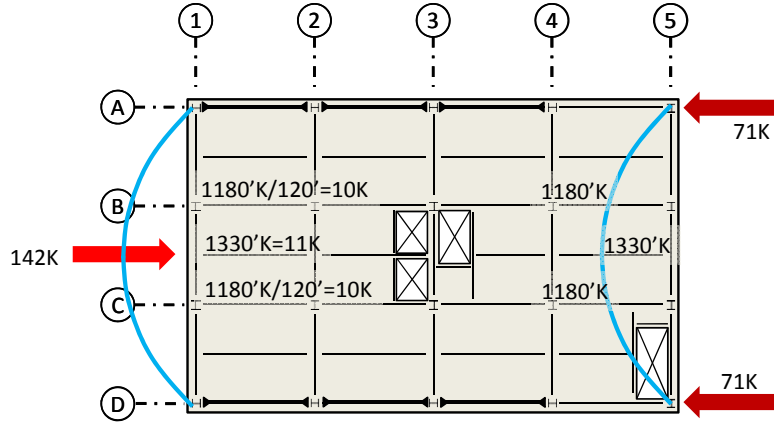
### Diaphragm analysis: Roof X



### Diaphragm analysis: Roof X



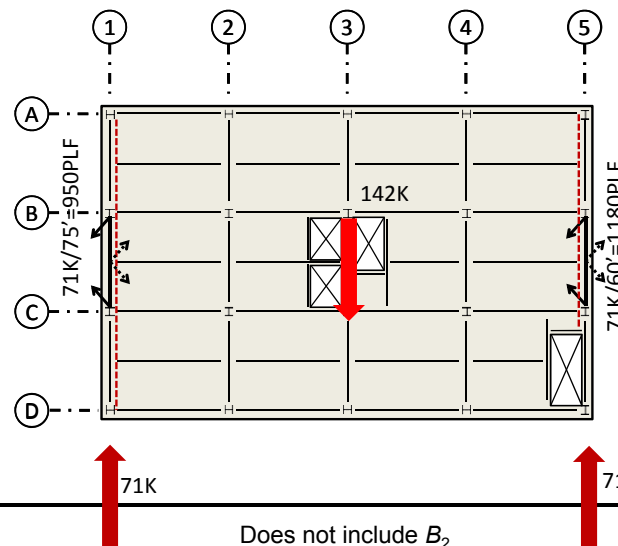
### Diaphragm analysis: Roof X



Does not include  $B_2$

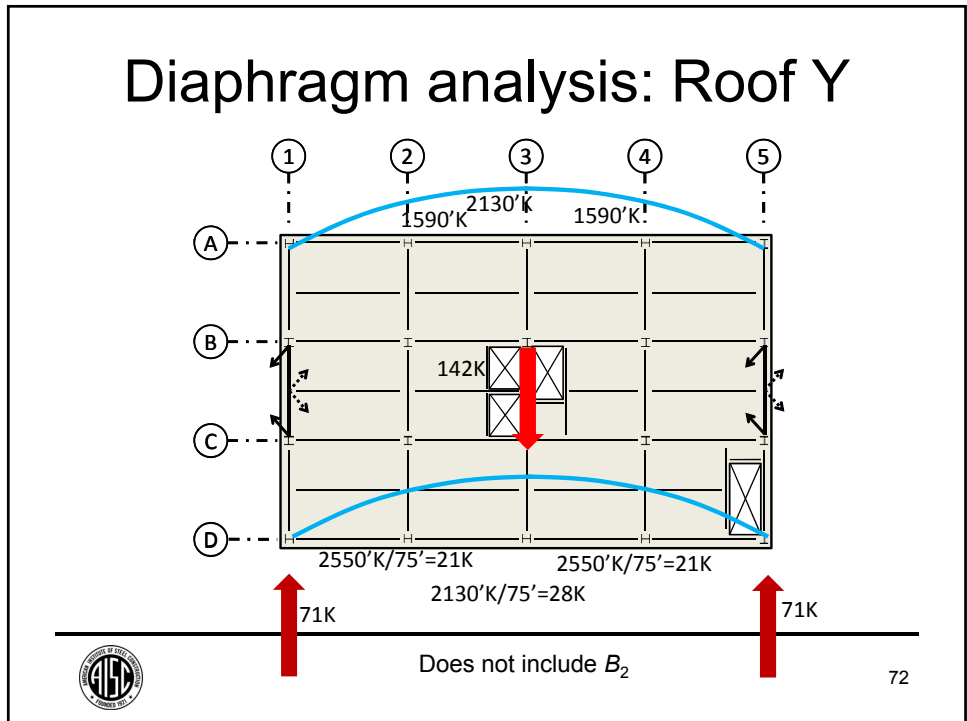
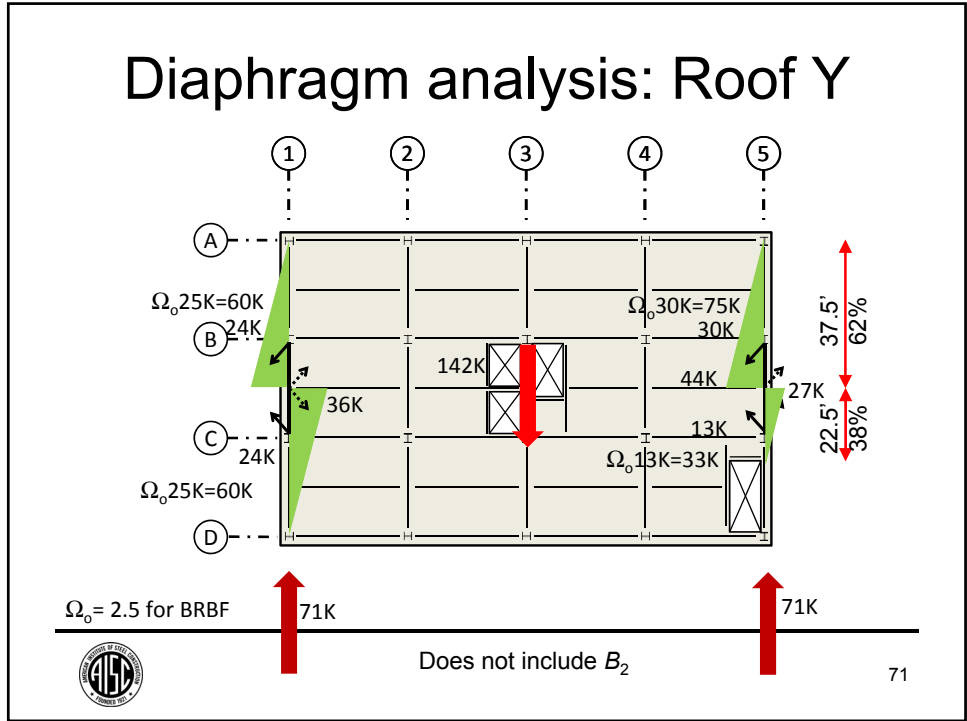
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### Diaphragm analysis: Roof Y

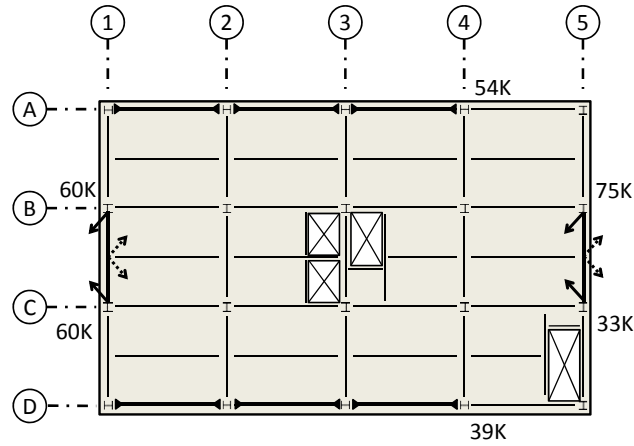


Does not include  $B_2$

70



## Chord/collector forces



Does not include  $B_2$

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## Capacity-design forces

There's always a solution in steel.



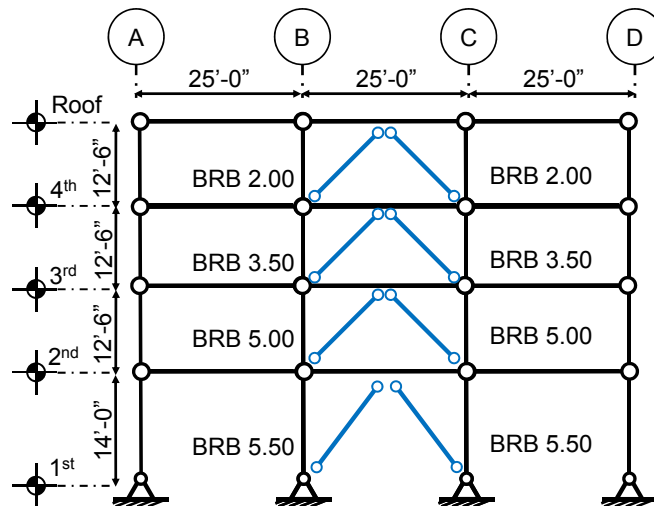
## Capacity-design forces

- Per ASCE 7 §12.4.3 the overstrength seismic load,  $\Omega_o E_h$ , need never be taken as greater than the capacity-limited seismic load effect, ( $E_{cl}$  in ASCE 7 2016)
- Capacity design can only happen after frames designed
- In this example we will show capacity-design forces prior to showing frame design



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## Brace sizes



Design of BRBF presented in Session 8

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## Brace capacity (4<sup>th</sup> floor)

- Tension
  - $\omega R_y F_{ySC} A$
  - $1.4(42\text{ksi})2.00\text{in}^2 = 118\text{K}$
- Compression
  - $\beta\omega R_y F_{ySC} A$
  - $1.15*1.4(42\text{ksi})2.00\text{in}^2 = 135\text{K}$
- Horizontal component:
  - $(118\text{K}+135\text{K})*\cos\theta = 179\text{K}$
  - $<B_2\Omega_o V_{frame} = 1.02*2.5*66\text{K} = 168\text{k}$
- Use capacity forces for roof collectors



AISC 341 §F4.2a

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## Diaphragm design

There's always a solution in steel.



## Deck selection

- Maximum shear
  - 1.18KLF
  - $B_2 * 1.18\text{KLF} = 1.20\text{KLF}$
- Design composite deck
  - Reinforced concrete section
  - Consider only topping above steel deck
    - 3.25" light weight concrete topping
    - #3 A614 Gr. 60 bars @12" each way



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## Deck selection

- Design composite deck
  - $v_c = 2\lambda d \sqrt{f'_c}$   
 $= (2)0.75(3.25")(4000)^{1/2} (12"/ft)$   
 $= 3700 \text{ plf}$
  - $v_s = A_s f_y$   
 $= 0.11\text{in}^2/\text{ft} * 60 \text{ ksi}$   
 $= 6600 \text{ plf}$
  - $\phi V_n = \phi (v_c + v_s)$   
 $= 0.75*(3700\text{plf}+6600\text{plf})$   
 $= 7725 \text{ plf} > 1200 \text{ plf}$



ACI 318 §11.2

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## Shear transfer

- Design for  $\Omega_o$  &  $B_2$ 
  - SMF
    - $1.02 \cdot 3 \cdot 0.644 \text{KLF} = 1.97 \text{KLF}$
  - BRBF
    - $2.5 \cdot 1.20 \text{KLF} = 3.00 \text{KLF}$
- Provide  $\frac{3}{4}$ "x4" studs @ 24" on collectors  
 $l_{s,req'd} = 2 \text{ in} + 1.5 \text{ in} = 3.5 \text{ in} < 4 \text{ in}$
- 4" stud projects above flute 2"



AISC 341 §B5.1

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## Shear transfer

- Stud strength:
  - $R_g = 1.0$  for one row with deck perpendicular (worst case)
 
$$Q_n = 0.5 A_{sc} \sqrt{f'_c E_c}$$

$$Q_n \leq R_g R_p A_{sc} F_u$$
  - $R_p = 0.6$  for one row with deck perpendicular (worst case)
 
$$A_{sc} = \frac{\pi}{4} \left( \frac{3}{4} \text{ in} \right)^2 = 0.44 \text{ in}^2$$

$$f'_c = 4000 \text{ psi}$$

$$E_c = w_c^{1.5} \sqrt{f'_c}$$

$$E_c = (115 \text{ pcf})^{1.5} \sqrt{4 \text{ ksi}}$$

$$E_c = 2466 \text{ ksi}$$
  - $F_u = 65 \text{ ksi}$



AISC 360 §18.2a

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## Shear transfer

- Stud strength:

$$Q_n = 0.5 \times 0.44 \text{ in}^2 \sqrt{4 \text{ ksi} \times 2466 \text{ ksi}} = 21.85 \text{ k}$$

$$Q_n \leq 1.0 \times 0.6 \times 0.44 \text{ in}^2 \times 65 \text{ ksi} = 17.2 \text{ k} \quad \text{Typically governs}$$

$$Q_n = 17.2 \text{ k}$$

$$\phi Q_n = 0.65 \times 17.2 \text{ k} = 11.2 \text{ k}$$

- Spaced @ 24"
  - 5.6 KLF > 3.0 KLF



AISC 360 §18.2a

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## Collector design

There's always a solution in steel.



## Collector design

- Combined flexure and axial
  - Compression governs over tension for collector member
    - Include  $P-\delta$  (in the form of  $B_1$ )
    - Perform Chapter H interaction
- Tension may govern for collector connection
  - Compression path through deck typically neglected



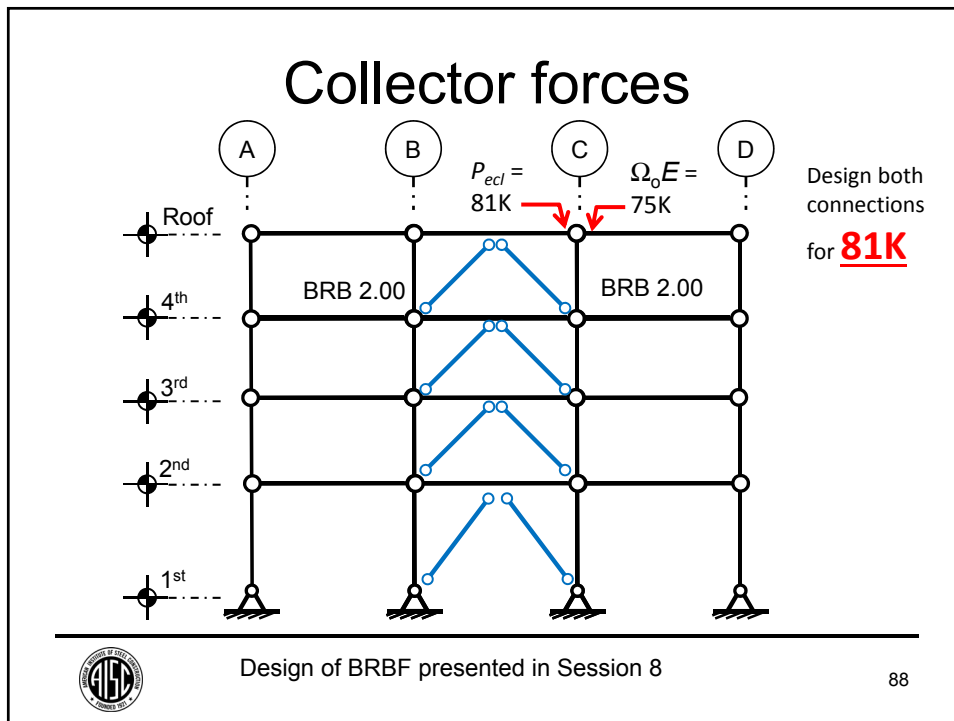
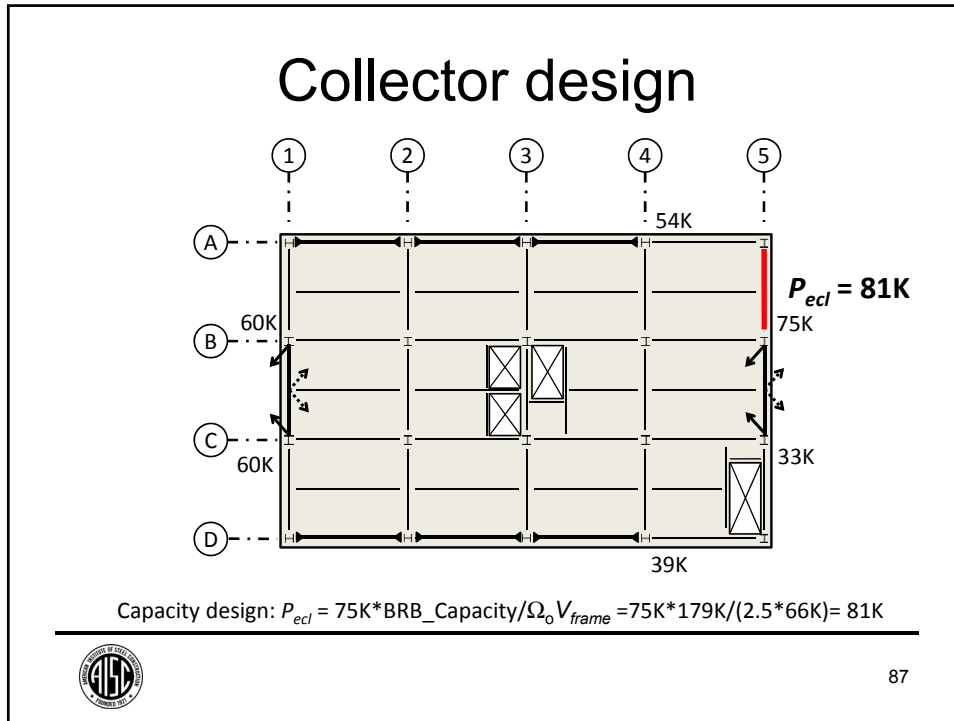
85

## Collector design

- Flexure
  - Composite strength
  - Continuously braced for LTB
- Compression
  - Flexural buckling
    - Major axis
    - Minor axis braced by composite deck
  - Torsional or flexural-torsional buckling
    - Twisting about restrained top flange



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# Collector Loads

- $M_u = 1.4D + 0.5L + E_{cl}$  combo CLLC-1
  - $M_u = 1.4(98.7'K) + 0.5(0'K) + 0'k = 138'K$
- $P_u = 1.4D + 0.5L + E_{cl}$  combo CLLC-1
  - $P_u = 1.4(0K) + 0.5(0K) + 81K = 81K$
- $V_u = 1.4D + 0.5L + E_{cl}$  combo CLLC-1
  - $V_u = 1.4(9.4K) + 0.5(0K) + 0K = 13.1K$
  - Beam shear design not presented
- Use W18x50 (per Seismic Design Manual)



Table 1-1 (continued) W-Shapes Dimensions													Table 1-1 (continued) W-Shapes Properties																
Shape	Area, A	Depth, d	Web		Flange		Distance		Work- able Cage	Compact Section Criteria			Axis X-X			Axis Y-Y			Torsional Properties										
			Thickness, t <sub>w</sub>	Width, b <sub>f</sub>	Thickness, t <sub>f</sub>	Width, b <sub>f</sub>	K <sub>xx</sub>	K <sub>yy</sub>		S <sub>x</sub>	I <sub>x</sub>	r <sub>x</sub>	S <sub>y</sub>	I <sub>y</sub>	r <sub>y</sub>	J	C <sub>w</sub>												
W21-x3	27.2	21.6	0.580	3/4	8.42	8 1/4	0.300	1/4	1.43	1/4	15 1/2	18 1/2	5 1/2	83	4.52	22.3	2070	192	870	221	82.8	22.1	1.84	34.7	2.24	20.7	6.03	8940	
x30	5.00	36.4	1830	171	8.67	196	81.4	19.5	1.83	30.5	2.21	20.6	4.34	8630															
x73 <sup>a</sup>	21.5	21.2	0.455	1/2	8.30	8 1/4	0.740	3/4	1.24	1 1/4	1/4	1/4	1/4	73	5.60	41.2	1800	151	864	172	70.6	17.0	1.81	26.6	2.19	20.6	3.02	7410	
x86 <sup>b</sup>	20.0	21.1	0.430	1/2	8.27	8 1/4	0.695	3/4	1.19	1 1/4	1/4	1/4	1/4	86	6.04	43.6	1480	140	860	160	64.7	15.7	1.80	24.4	2.17	20.4	2.45	6760	
x95 <sup>c</sup>	18.3	21.0	0.400	1/2	8.24	8 1/4	0.615	3/4	1.12	1 1/4	1/4	1/4	1/4	92	6.70	46.9	1320	127	854	144	57.5	14.0	1.77	21.7	2.15	20.4	1.93	5960	
x55 <sup>d</sup>	16.2	20.8	0.375	1/2	8.22	8 1/4	0.522	1/2	1.02	1 1/4	1/4	1/4	1/4	55	7.87	50.0	1140	110	840	126	48.4	11.8	1.73	18.4	2.11	20.3	1.24	4980	
x40 <sup>e,f</sup>	14.1	20.6	0.350	1/2	8.14	8 1/4	0.430	1/2	0.930	1 1/4	1/4	1/4	1/4	48	9.47	53.6	959	93.0	824	107	38.7	9.52	1.66	14.9	2.05	20.2	0.903	3950	
W21-x57 <sup>g</sup>	16.7	21.1	0.405	1/2	8.36	8 1/4	0.650	3/4	1.15	1 1/4	1/4	1/4	1/4	57	5.04	46.3	1170	111	836	129	30.6	9.35	1.36	14.8	1.60	20.5	1.77	3190	
x50 <sup>h</sup>	14.7	20.8	0.380	1/2	8.33	8 1/4	0.535	3/4	1.04	1 1/4	1/4	1/4	1/4	50	6.16	49.4	964	94.5	816	110	24.9	7.64	1.30	12.2	1.64	20.3	1.14	2570	
x44 <sup>i</sup>	13.0	20.7	0.350	1/2	8.00	8 1/4	0.450	1/2	0.950	1 1/4	1/4	1/4	1/4	44	7.22	53.6	843	81.6	806	95.4	20.7	6.37	1.26	10.2	1.60	20.3	0.770	2110	
W18-x11 <sup>j</sup>	91.6	22.3	2.23 <sup>k</sup>	1 1/2	12.0	12	2.74	2 1/4	3.24	3 1/4	1 1/4	1 1/4	1 1/4	311	2.18	10.4	6970	624	872	754	795	132	2.95	20.7	3.51	19.6	176	76200	
x20 <sup>l</sup>	83.3	21.9	2.14 <sup>k</sup>	1 1/2	11.9	11 1/2	2.50	2 1/4	3.00	3 1/4	1 1/4	1 1/4	1 1/4	283	2.36	11.3	6170	565	861	676	704	118	2.91	18.5	3.47	19.4	134	65900	
x25 <sup>m</sup>	76.0	21.5	2.10 <sup>k</sup>	1 1/2	11.8	11 1/4	2.30	2 1/4	2.70	3	1 1/4	1 1/4	1 1/4	258	2.56	12.5	5510	514	853	611	628	107	2.88	18.6	3.42	19.2	103	57600	
x29 <sup>n</sup>	68.6	21.1	1.16	1 1/4	11.7	11 1/4	2.11	2 1/4	2.51	2 1/4	1 1/4	1 1/4	1 1/4	234	2.76	13.8	4900	466	844	449	558	96.8	2.85	14.9	3.27	19.0	78.7	50100	
x33 <sup>o</sup>	62.3	20.7	2.05 <sup>k</sup>	1 1/2	11.6	11 1/4	1.91	1 1/4	2.31	2 1/4	1 1/4	1 1/4	1 1/4	211	3.02	15.1	4330	419	835	490	493	85.3	2.82	13.2	3.32	18.8	58.6	43400	
x39 <sup>p</sup>	56.2	20.4	2.05 <sup>k</sup>	1 1/2	11.5	11 1/4	1.75	1 1/4	2.15	2 1/4	1 1/4	1 1/4	1 1/4	192	3.27	16.7	3870	380	828	442	440	76.8	2.79	11.9	3.28	18.7	44.7	39000	
x45 <sup>q</sup>	51.4	20.0	2.00	1 1/2	11.4	11 1/4	1.59	1 1/4	1.99	2 1/4	1 1/4	1 1/4	1 1/4	175	3.58	18.0	3450	344	820	399	391	68.8	2.76	10.6	3.26	18.4	33.6	33300	
x50 <sup>r</sup>	46.3	19.7	1.87 <sup>k</sup>	1 1/4	11.3	11 1/4	1.44	1 1/4	1.84	2 1/4	1 1/4	1 1/4	1 1/4	159	3.95	19.8	3000	310	812	356	347	61.4	2.74	9.48	3.20	18.3	25.2	29000	
x54 <sup>s</sup>	42.0	19.5	1.95 <sup>k</sup>	1 1/4	11.2	11 1/4	1.32	1 1/4	1.72	2 1/4	1 1/4	1 1/4	1 1/4	143	4.25	22.0	2750	282	809	322	311	55.5	2.72	8.54	3.17	18.2	19.2	25700	
x60 <sup>t</sup>	38.0	19.3	1.95 <sup>k</sup>	1 1/4	11.2	11 1/4	1.20	1 1/4	1.60	2 1/4	1 1/4	1 1/4	1 1/4	130	4.62	23.9	2460	256	803	290	278	49.9	2.70	7.67	3.13	18.1	14.5	22700	
x66 <sup>u</sup>	35.1	19.0	1.95 <sup>k</sup>	1 1/4	11.3	11 1/4	1.08	1 1/4	1.46	1 1/4	1 1/4	1 1/4	1 1/4	119	5.31	24.5	2190	231	750	282	253	44.9	2.69	6.81	3.10	17.9	10.6	20300	
x72 <sup>v</sup>	31.1	18.7	1.87 <sup>k</sup>	1 1/4	11.2	11 1/4	0.940 <sup>k</sup>	1 1/4	1.34	1 1/4	1 1/4	1 1/4	1 1/4	106	5.96	27.2	1910	204	784	230	220	39.4	2.66	6.05	3.10	17.8	7.48	17400	
x78 <sup>w</sup>	28.5	18.6	1.87 <sup>k</sup>	1 1/4	11.1	11 1/4	0.870 <sup>k</sup>	1 1/4	1.27	1 1/4	1 1/4	1 1/4	1 1/4	97	6.41	30.0	1750	188	782	211	201	36.1	2.65	5.53	3.08	17.7	5.86	15800	
x84 <sup>x</sup>	25.3	18.4	1.87 <sup>k</sup>	1 1/4	11.1	11 1/4	0.770 <sup>k</sup>	1 1/4	1.17	1 1/4	1 1/4	1 1/4	1 1/4	86	7.22	33.4	1530	166	777	186	175	31.6	2.63	4.84	3.03	17.6	4.10	13600	
x90 <sup>y</sup>	22.3	18.2	1.87 <sup>k</sup>	1 1/4	11.0	11	0.680 <sup>k</sup>	1 1/4	1.08	1 1/4	1 1/4	1 1/4	1 1/4	76	8.11	37.8	1300	146	773	163	152	27.6	2.61	4.22	3.02	17.5	2.83	11700	
W18-x71 <sup>z</sup>	20.9	18.5	1.87 <sup>k</sup>	1 1/4	7.64	7 1/4	0.810 <sup>k</sup>	1 1/4	1.21	1 1/4	1 1/4	1 1/4	1 1/4	71	4.71	32.4	1170	127	750	146	60.3	15.8	1.70	24.7	2.05	17.7	3.49	4700	
x65	19.1	18.4	1.87 <sup>k</sup>	1 1/4	7.59	7 1/4	0.750 <sup>k</sup>	1 1/4	1.15	1 1/4	1 1/4	1 1/4	1 1/4	65	5.06	35.7	1070	117	749	133	54.8	14.4	1.69	22.5	2.01	17.7	2.73	4240	
x60 <sup>aa</sup>	17.6	18.2	1.87 <sup>k</sup>	1 1/4	7.56	7 1/4	0.695 <sup>k</sup>	1 1/4	1.10	1 1/4	1 1/4	1 1/4	1 1/4	60	5.44	38.7	964	108	747	123	50.1	13.3	1.68	20.6	2.02	17.5	2.17	3650	
x55 <sup>ab</sup>	16.2	18.1	1.87 <sup>k</sup>	1 1/4	7.50	7 1/4	0.630 <sup>k</sup>	1 1/4	1.03	1 1/4	1 1/4	1 1/4	1 1/4	55	5.98	41.1	806	88.3	741	112	44.8	11.9	1.67	18.5	2.00	17.5	1.86	3430	
x50 <sup>ac</sup>	14.7	18.0	1.8	1 1/4	7.50	7 1/4	0.570 <sup>k</sup>	1 1/4	0.972 <sup>k</sup>	1 1/4	1 1/4	1 1/4	1 1/4	50	6.57	45.2	800	88.9	738	101	40.1	10.7	1.65	16.6	1.98	17.4	1.24	3040	
W18-x40 <sup>ad</sup>	13.5	18.1	1.8	1 1/4	6.06	6	0.605 <sup>k</sup>	1 1/4	1.01	1 1/4	1 1/4	1 1/4	1 1/4	46	5.01	44.6	712	78.8	725	90.7	22.5	7.43	1.29	11.7	1.58	17.5	1.22	1720	
x40 <sup>ae</sup>	11.8	17.9	1.77 <sup>k</sup>	1 1/4	6.02	6	0.525 <sup>k</sup>	1 1/4	0.927 <sup>k</sup>	1 1/4	1 1/4	1 1/4	1 1/4	40	5.73	50.9	612	68.4	721	78.4	18.1	6.35	1.27	10.0	1.58	17.4	0.810	1440	
x35 <sup>af</sup>	10.3	17.7	1.77 <sup>k</sup>	1 1/4	6.00	6	0.425 <sup>k</sup>	1 1/4	0.827 <sup>k</sup>	1 1/4	1 1/4	1 1/4	1 1/4	35	7.06	53.5	510	57.6	704	66.5	15.3	5.12	1.22	8.06	1.51	17.3	0.508	1140	

<sup>a</sup> Shape is slender for compression with F<sub>y</sub> = 50 ksi.  
<sup>b</sup> Shape exceeds compact limit for flexure with F<sub>y</sub> = 50 ksi.  
<sup>c</sup> The actual size, composition and orientation of fastener components should be compared with the geometry of the cross section to ensure compatibility.  
<sup>d</sup> Flange thickness greater than 2 in. Special requirements may apply per AISI Specification Section A3.1c.



## Collector design

- Use W18x50 (from Seismic Design Manual Example 8.4.1)
  - Member properties (units per Manual)

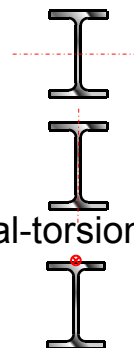
W18x50							
$A$	$d$	$t_w$	$b_f$	$t_f$	$k_1$	$b/2t_f$	$h/t_w$
14.7	18.0	0.355	7.50	0.570	13/16	6.57	45.2
$I_x$	$Z_x$	$S_x$	$r_x$	$h/t_w > 1.49 \sqrt{E/F_y} = 35.9$			
800	101	88.9	7.38				
$I_y$	$Z_y$	$S_y$	$r_y$	Web is not compact	$J$	$C_w$	
40.1	16.6	10.7	1.65		1.24	3040	



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## Collector design

- Compressive Strength
  - Major axis
    - $(KL)_x = 25'-0''$
  - Minor axis
    - $(KL)_y = 0'-0''$
  - Constrained-axis flexural-torsional buckling
    - $(KL)_{CAFT} = 12'-6''$



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## Collector design

- Compressive Strength
  - Major axis buckling
    - $KL/r_x = 300"/7.38" = 40.7$
    - From Table 4-22:  $KL/r=41$ :  $\phi F_{cr} = 39.8\text{ksi}$ 
      - $\phi F_{cr} A = 39.8\text{ksi} (14.7\text{in}^2) = 585\text{K}$



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## Collector design

- Compressive Strength
  - CAFT buckling
    - $F_e = 0.9 \left[ \frac{\pi^2 E [C_w + I_y (d/2)^2]}{(K_2 L)^2} + GJ \right] \frac{1}{I_x + I_y + (d/2)^2 A_g}$
    - AISC EJ (2013 Q4)
      - differs by factor of 0.9 from 2<sup>nd</sup> Edition Seismic Design Manual
    - $F_e = 0.9(46.2\text{ksi}) = 41.6\text{ksi}$
    - $QF_y / F_e = 1(50\text{ksi})/41.6\text{ksi} = 1.20 < 2.25$



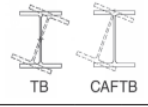
Torsional and Constrained-Axis Flexural-Torsional Buckling  
 Tables for Steel W-Shapes in Compression

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## Collector design

$F_y = 50$  ksi

Table 1. (continued)  
 Torsional Buckling Design Strength in Axial Compression  
 $\phi_c P_n$ , kip



Shape	W18x								W16x					
	50		46		40		35		100		89		77	
10	509	465	435	366	361	306	301	252	1190	1140	1050	1010	898	864
11	492	439	418	336	345	280	286	229	1170	1110	1030	981	880	838
12	475	413	401	308	329	255	271	205	1150	1080	1010	953	863	812
13	459	386	386	281	$P_u / \phi P_n = 81K / 386K = 0.21$									
14	443	360	371	257										
15	428	335	357	235										
16	414	311	344	216	276	170	217	131	1090	969	948	843	796	709
17	399	289	332	199	264	156	205	120	1070	943	934	818	781	684
18	385	269	322	185	254	145	195	110	1060	918	920	793	767	660
19	373	249	313	174	244	135	185	102	1050	895	908	770	754	638
20	361	232	304	164	235	126	175	95.1	1040	873	896	748	742	616
22	340	204	289	148	220	113	160	84.0	1020	832	876	706	719	575



Torsional and Constrained-Axis Flexural-Torsional Buckling Tables for Steel W-Shapes in Compression

## Collector Design

- Required flexural strength
  - $P-\delta$  amplification
    - $P_E = \pi^2 EI / (KL_x)^2 = 2540K$
    - $B_1 = \frac{C_m}{1 - P_u/P_E} = \frac{1}{1 - 81K/2540K} = 1.03$
    - $M_u = 138'K * 1.03 = 142'K$



# Collector Design

- Flexural strength
  - AISC Manual Table 3-19
    - $Y_2=3.5\text{in}$
    - Use  $\Sigma Q_n=184\text{K}$ 
      - $184\text{K}/11.2\text{K}/_{\text{stud}}=16.4$  studs (each side of midpoint)
      - 32.8 studs
  - Collector studs
    - $81\text{K}/11.2\text{K}/_{\text{stud}}=7.2$  studs
    - Not additive to flexure studs
- $\phi M_n = 516$  kip-ft

**steelwise**

**Under Foot**

BY SUSAN BURMEISTER, P.E., AND WILLIAM P. JACOBS, P.E.

Horizontal floor diaphragm load effects on composite beam design.

DECEMBER 2008 MODERN STEEL CONSTRUCTION



**W18**

**Table 3-19 (continued)**  
**Composite W-Shapes**  
 Available Strength in Flexure,  
 kip-ft

$F_y = 50$  ksi

Shape	$M_p/\Omega_b$ $\phi_b M_p$		PNA <sup>c</sup>	$Y_1^a$ in.	$\Sigma Q_n$ kip	$Y_2^b$ , in.							
	kip-ft					2		2.5		3		3.5	
	ASD	LRFD				ASD	LRFD	ASD	LRFD	ASD	LRFD	ASD	LRFD
W18x50	252	379	TFL	0	735	403	606	422	634	440	662	458	689
			2	0.143	628	392	590	408	613	424	637	439	660
			3	0.285	521	381	572	394	592	407	611	420	631
			4	0.428	414	368	553	378	569	389	584	399	600
			BFL	0.570	308	355	533	362	545	370	556	378	568
			6	2.08	246	345	518	351	527	357	537	363	546
			7	3.82	184								
<b>ASD</b>	<b>LRFD</b>	<sup>a</sup> $Y_1$ = distance from top of the steel beam to plastic neutral axis <sup>b</sup> $Y_2$ = distance from top of the steel beam to concrete flange force <sup>c</sup> See Figure 3-3c for PNA locations.											
$\Omega_b = 1.67$	$\phi_b = 0.90$												



## Collector design

- Try non-composite flexural strength
  - Manual Table 3-2
  - $\phi M_n = 379$  kip-ft
- $P_u / \phi P_n = \underline{81K / 386K} = \underline{0.21} > 0.2$
- $P_u / \phi P_n + \frac{8}{9} M_u / \phi M_n$  (H1-1b)
- $= (\underline{0.21}) + \frac{8}{9} (142'K) / (379'K) = \underline{0.55}$  OK
  - Provide studs @ 24" (12 studs)



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## Collector connections

There's always a solution in steel.



## Collector connection

- Single-plate connection
  - (4)  $\frac{7}{8}$ "  $\varnothing$  A325N bolts
    - 3" spacing
    - 1.5" edge distance top & bottom
    - 2.5" side edge distance
  - $\frac{3}{8}$ " A36 plate
    - $\frac{1}{4}$ " double-sided fillets

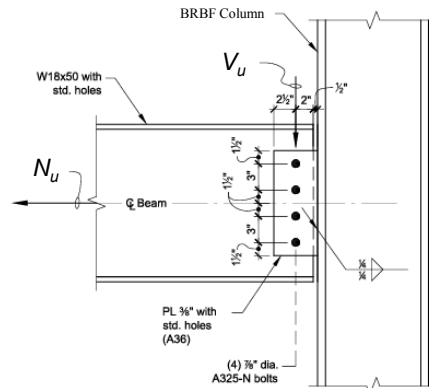


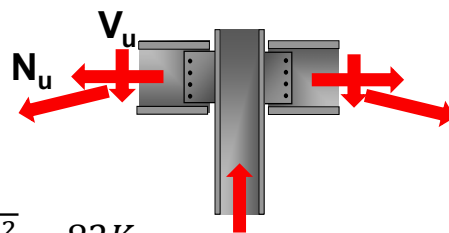
Fig. 8-6. Collector connection investigated in Example 8.4.2.



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## Required strength

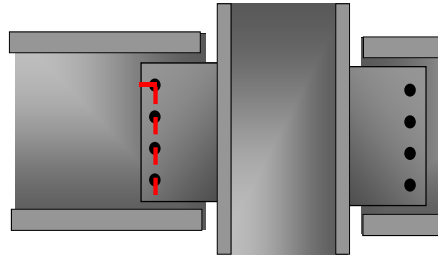
- $V_u = 13K$
- $N_u = 81K$
- $R_u = \sqrt{V_u^2 + N_u^2}$ 
  - $= \sqrt{(13K)^2 + (81K)^2} = 82K$
- $\theta = \tan^{-1} \frac{13K}{81k} = 9^\circ$



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## Collector connection

- Alternative approaches
  - Evaluate as a load at  $90^\circ$ 
    - This approach shown in SDM
  - Evaluate shear and tension separately
    - SRSS interaction
    - This approach taken here



- Shear strength
  - From Table 10-10a
    - $\phi V_n = 78.3K$



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**Table 10-10a (continued)**  
**Single-Plate Connections**  
 Bolt, Weld and Single-Plate Available Strengths, kips

Plate  $F_y = 36$  ksi **7/8-in.-diameter bolts**

n	Bolt Group	Thread Cond.	Hole Type	Plate Thickness, in.											
				1/4		5/16		3/8		7/16		1/2		9/16	
				ASD	LRFD	ASD	LRFD	ASD	LRFD	ASD	LRFD	ASD	LRFD	ASD	LRFD
4 (L = 12)	Group A	N	STD	34.9	52.2	43.5	65.3	52.2	78.3	60.9	91.4	69.6	104	---	---
			SSLT	34.9	52.2	43.5	65.3	52.2	78.3	60.9	91.4	69.6	104	71.2	107
		X	STD	34.9	52.2	43.5	65.3	52.2	78.3	60.9	91.4	69.6	104	---	---
			SSLT	34.9	52.2	43.5	65.3	52.2	78.3	60.9	91.4	69.6	104	71.2	107
	Group B	N	STD	34.9	52.2	43.5	65.3	52.2	78.3	60.9	91.4	69.6	104	---	---
		SSLT	34.9	52.2	43.5	65.3	52.2	78.3	60.9	91.4	69.6	104	71.2	107	

$$\sqrt{\left(\frac{V_u}{\phi V_n}\right)^2 + \left(\frac{T_u}{\phi T_n}\right)^2} \leq 1 \quad \frac{T_u}{\phi T_n} \leq \sqrt{1 - \left(\frac{V_u}{\phi V_n}\right)^2} = 0.98$$

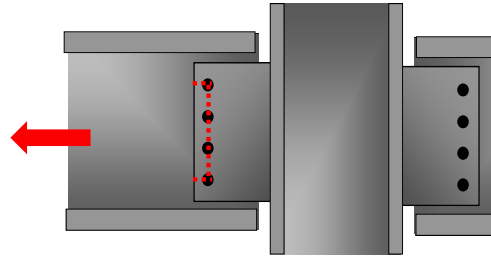
$$\phi T_n \geq T_u / 0.98 \quad \phi T_n \geq 81K / 0.98 = 82K$$



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## Collector connection

- Tension strength
  - Bolt (AISC 360 §J3)
    - Shear
    - Bearing
    - Tearout
  - Plate (AISC 360 §J4)
    - Yield
    - Rupture
    - Block shear
  - Weld (AISC 360 §J2)
    - Exceeds plate strength



- Beam block shear, bearing, tearout
  - $t_w(65\text{ksi}/58\text{ksi})=$ 
    - $0.355''(65/58)=0.39'' > \frac{3}{8}''$
  - $t_w(50\text{ksi}/36\text{ksi})=$ 
    - $0.355''(50/36)=0.49'' > \frac{3}{8}''$
  - Plate governs



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## Collector connection

- Plate limit states
  - Yield:  $\phi F_y A$ 
    - $= 0.9(36\text{ksi})(\frac{3}{8}'')(12'')$
    - $= 146\text{K}$
  - Rupture:  $\phi F_u A_e = F_u A_n$ 
    - $= 0.75(58\text{ksi})(\frac{3}{8}'')(12''-4'')$
    - $= 130\text{K}$
- Block shear  $\phi R_n \leq$ 
  - $\phi(0.6F_u A_{nv} + U_{BS}F_u A_{nt})$ 
    - $0.75(\frac{3}{8}'')(0.6*58\text{ksi}*2*2'' + 1.0*58\text{ksi}*6'')= 137\text{K}$
  - $\phi(0.6F_y A_{gv} + U_{BS}F_u A_{nt})$ 
    - $0.75(\frac{3}{8}'')(0.6*50\text{ksi}*2*2.5'' + 1.0*58\text{ksi}*6'')= 140\text{K}$



AISC 360 §J4

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## Collector connection

- Bolt limit states
  - Shear: Table 7-1
    - $4 \times 24.3K = 97.2K$
  - Bearing (spacing)
    - Table 7-4
    - $4(\frac{3}{8})91.4K/in = 137K$
  - Bearing (edge distance)
    - Table 7-5
    - $4(\frac{3}{8})79.9/in = 120K$
- Governing strength:
  - $\phi R_n = 97.2K$
  - $R_u / \phi R_n = 81K / 97K = 0.84$

$$\sqrt{\left(\frac{V_u}{\phi V_n}\right)^2 + \left(\frac{T_u}{\phi T_n}\right)^2} = \sqrt{(0.18)^2 + (0.84)^2} = 0.85 \quad OK$$



AISC 360 §J3

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## Diaphragm openings

There's always a solution in steel.



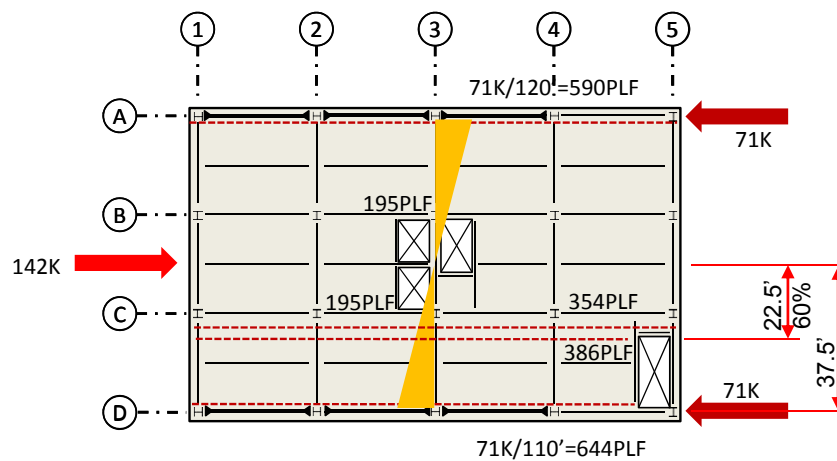
## Diaphragm openings

- Local shears
- Local collector forces
- Local chord forces



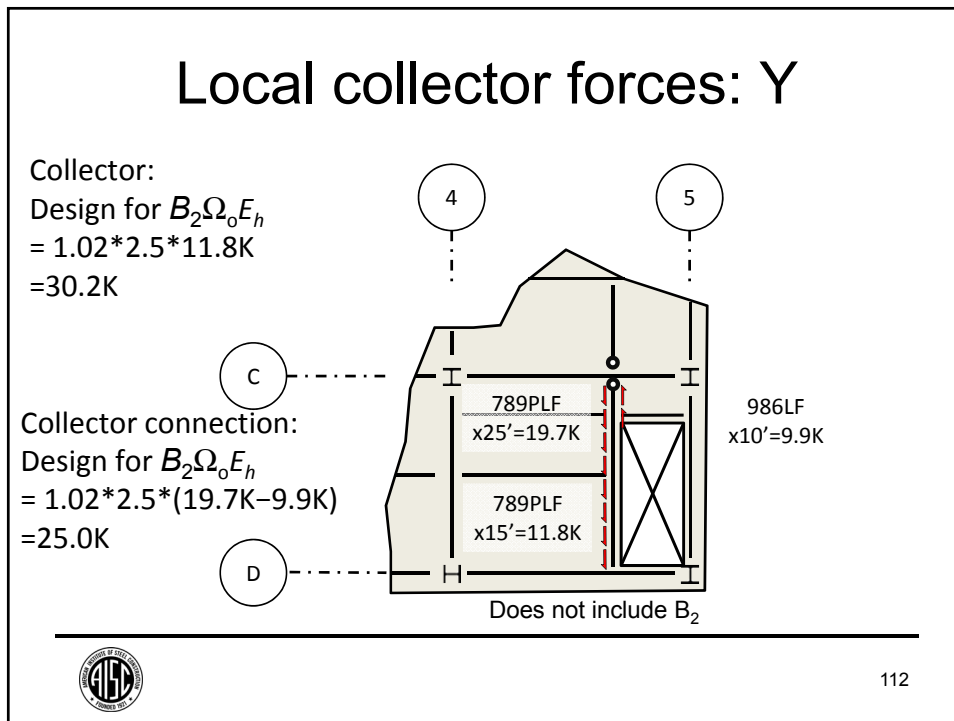
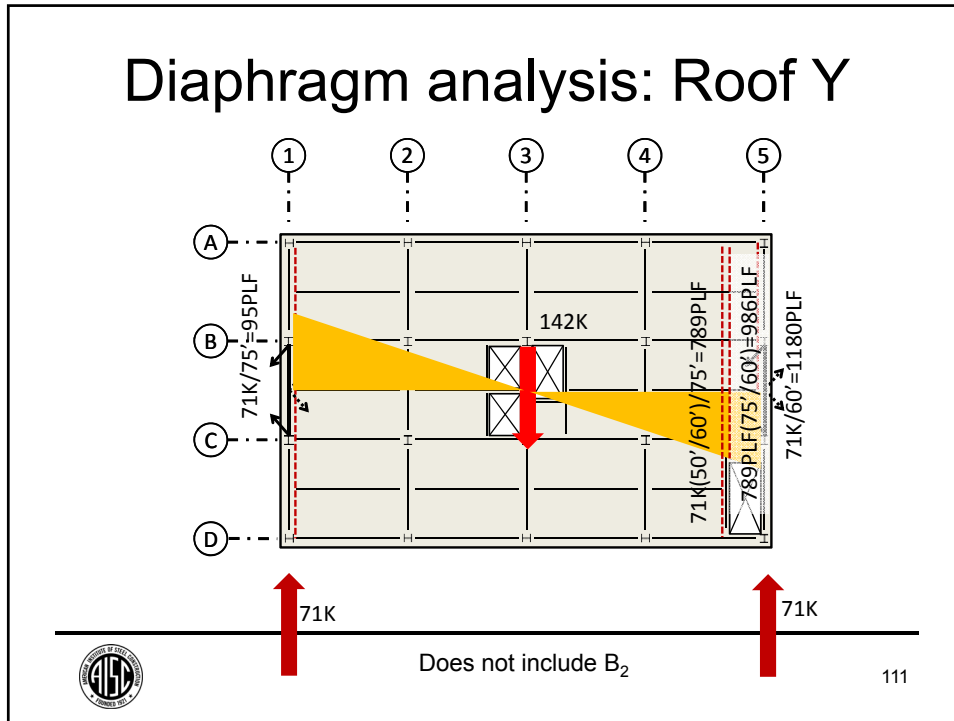
109

## Diaphragm analysis: Roof X



Does not include B<sub>2</sub>

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### Local chord forces: Y

$$\frac{1.18klf + 1.08klf}{2} 10'$$

= 11.3K

Chord and connection:  
 Design for  $B_2E_h$   
 =  $1.02 \cdot 11.3K$   
 = 11.5K

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There's always a solution in steel.

## Summary

## Summary

- Simple design methods presented
- Methods of accounting for second-order effects presented
- Forces generated for design of SMF & BRBF
- Diaphragm forces generated
- Roof diaphragm analyzed
- Deck designed
- Example collector designed
- Example collector connection designed



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## End of session 6

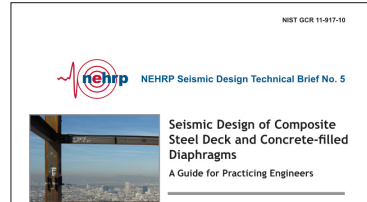
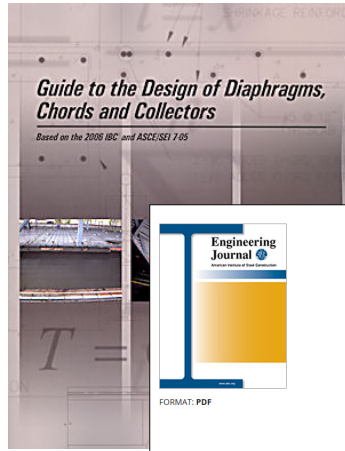
*Next:*

## Design of the Moment frames

There's always a solution in steel.



## Additional resources



**Engineering Journal**  
 Torsional and Constrained-Axis Flexural-Torsional Buckling Tables for Steel W-Shapes in Compression

MEMBER	NON-MEMBER
FREE	\$10.00

1 ADD TO CART

FORMAT: PDF

Liu, D.; Davis, B.; Arber, L.; Sabelli, R. (2013). "Torsional and Constrained-Axis Flexural-Torsional Buckling Tables for Steel W-Shapes in Compression." *Engineering Journal*, American Institute of Steel Construction, Vol. 50, pp. 205-247.

Torsional buckling (TB), an applicable limit state for W-shape members subject to axial compression, often controls when the torsional effective unbraced length exceeds the minor-axis flexural buckling effective unbraced length. Constrained-axis flexural-torsional buckling (CAFTB) is a potential limit state for W-shape members that are constrained to buckle with the center of twist at a location other than the centroidal axis, as is the case for a typical beam with one flange braced by a diaphragm and the other unbraced. Manual calculation of the TB or CAFTB available compressive strength is a somewhat lengthy process, especially when the section is slender for axial compression, and no design aid currently exists in the AISC Manual. This paper provides tables that facilitate the determination of TB and CAFTB available compressive strengths. Several example calculations are also provided.

Published: 2013, Quarter 4

AUTHOR(S)  
 Di Liu, Brad Davis, Leigh Arber, and Rafael Sabelli



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